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**Pacific Northwest  
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## Overview of the Flammability of Gases Generated in Hanford Waste Tanks

L. A. Mahoney  
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G. D. Johnson

July 2000



Prepared for the U.S. Department of Energy  
under Contract DE-AC06-76RLO 1830

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## Overview of the Flammability of Gases Generated in Hanford Waste Tanks

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Pacific Northwest National Laboratory  
Richland, Washington 99352

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(a) CH2M Hill Hanford Group, Inc.

## Executive Summary

This report presents an overview of what is known about the flammability of the gases generated and retained in Hanford waste tanks in terms of the gas composition, the flammability and detonability limits of the gas constituents, and the availability of ignition sources. The intrinsic flammability (or nonflammability) of waste gas mixtures is one major determinant of whether a flammable region develops in the tank headspace; other factors are the rate, surface area, volume of the release, and the tank ventilation rate, which are not covered in this report.

The flammability hazard in Hanford waste tanks was first recognized in the large periodic gas releases that occurred in the double-shell Tank 241-SY-101 (SY-101). In the early 1990s in that tank, periodic spontaneous gas releases resulted when gas that was generated accumulated in a layer of settled solids and made parts of the layer buoyant, causing those regions to break away, rise, and release part of their gas contents to the surface. These buoyant displacement gas release events (BD GREs) are possible only in tanks with deep supernatant layers. This currently includes only double-shell tanks. There are no deep supernatant layers in single-shell tanks, and as a result release events in these tanks have characteristically been small, produced by waste-disrupting activities (such as salt-well pumping) and very localized spontaneous mechanisms.

Waste in the Hanford tanks is generally classified into nonconvective and convective layers. Nonconvective layers contain enough solids to prevent temperature-gradient-driven convection, as shown by nonuniform temperature profiles. Convective layers have uniform temperature profiles and are sufficiently liquid to allow convection. They are called supernatant layers if they lie above a nonconvective layer. In common parlance, nonconvective layer refers to the settled solids on the bottom of a tank or to exceptionally thick floating layers of solids such as those in Tanks A-101 and AX-101. Relatively thin solids layers that float on a convective layer are called crusts. Nonconvective layers can be submerged beneath a supernatant layer or exposed to the headspace, as after salt-well pumping, wherein the upper part of the layer contains little liquid.

Flammable gases are generated in the liquid waste and retained in substantial quantities only in the nonconvective and crust layers. In SY-101, the mechanically mixed slurry layer also contained considerable gas. The flammability of the gases becomes a concern only after large releases to the headspace or to other air-filled volumes such as drill strings. Steady-state releases from the waste produce background concentrations that have been far below flammability. Gases stored in nonconvective layers and crusts are present in isolated bubbles or local networks that are too small to support in situ combustion and lack ignition sources. Though pore networks in the dry upper part of salt-well pumped waste may be large in extent, gases generated in the wet waste below diffuse so rapidly through the pores that flammable concentrations cannot exist there.

The gases that are of most concern in flammability assessment are hydrogen ( $H_2$ ), a highly flammable fuel that makes up a large part of the gas in many of the sampled tanks; ammonia ( $NH_3$ ), a less flammable and less prevalent fuel; and nitrous oxide ( $N_2O$ ), an oxidant (at sufficiently high temperature). Methane ( $CH_4$ ) and other hydrocarbons are highly flammable fuels but present only in negligible concentrations. Nitrogen ( $N_2$ ) is an inert gas that tends to

stifle flammability; water vapor ( $\text{H}_2\text{O}$ ), is also an inerting agent but present in smaller concentrations than nitrogen. Oxygen ( $\text{O}_2$ ) is the primary oxidant in the headspace but is not available within the waste. The relative amounts of fuel, oxidant, and inert gas in the waste in a given tank determine the gas flammability. The relative accuracies of the different types of gas composition measurements are important.

A review of the waste gas composition data from headspace samples, retained gas sampler (RGS) samples, drill string samples, and laboratory gas generation studies showed that

- In some cases (AW-101, A-101, and AN-103) the  $\text{H}_2/\text{N}_2\text{O}$  ratios measured by RGS are substantially different for the convective and nonconvective layers. The direction of the difference is not consistent.
- In almost all cases (S-111 and SY-101 are exceptions), the RGS-measured  $\text{H}_2/\text{N}_2$  ratio in the gas retained in the convective layer is between 0.2 and 0.45.
- The  $\text{H}_2/\text{N}_2\text{O}$  and  $\text{H}_2/\text{N}_2$  ratios of laboratory-generated gases do not consistently match those in gases measured in the waste or in the headspace. The direction of the difference is not consistent.
- The overall composition ( $\text{H}_2/\text{N}_2\text{O}/\text{N}_2$ ) of retained gas cannot be determined reliably from drill string samples because the  $\text{H}_2/\text{N}_2$  ratio cannot be measured meaningfully in high-air samples and the  $\text{H}_2/\text{N}_2\text{O}$  ratio does not consistently match RGS values.

The in situ gas composition (plus evaporated ammonia) is most appropriate for assessing the large sudden releases, while the headspace gas composition is appropriate for steady-state conditions. RGS composition data, where available, are preferred to define in situ gas composition because the measurements were made as nearly in situ as possible and contain all the constituents that affect flammability (Barker et al. 1999). However, ammonia released by evaporation should be added to RGS compositions of retained gas to represent the total release composition.

Because the released gas is highly diluted with air, drill string and headspace data lack  $\text{N}_2$  measurements, which, in most RGS samples, made up 20 to 50 mol% of the gas. It follows that  $\text{N}_2\text{O}$  and  $\text{H}_2$  together usually made up 50 to 80 mol% of the gas, so the absence of  $\text{N}_2$  data introduces about 25% uncertainty into estimates of the  $\text{H}_2$  flammability contribution in the headspace after a large release. A larger uncertainty comes from the fact that the  $\text{H}_2/\text{N}_2\text{O}$  ratios from headspace and drill string measurements often were within about 50% of the RGS values but could as easily differ by more than a factor of two. Drill string  $\text{H}_2/\text{N}_2\text{O}$  ratios were usually closer to RGS values than were the headspace measurements.

The flammability limits of a combustible mixture are those limiting compositions that will just support flame propagation when stimulated by an external ignition source. Flame propagation in fuel-rich mixtures is limited by oxidant availability and in lean mixtures by fuel availability. Thus the conditions of flammability are characterized by the upper (rich) and lower (lean) flammability limits of fuel gas concentration. In considering the flammability hazard associated with tank wastes, we are concerned more with the lower flammability limit (LFL) than with the upper flammability limit (UFL). The LFL is determined by the gas composition,

energy and location of the ignition source, direction of flame propagation relative to gravity, and whether the gas is quiescent or being mixed.

Using a large set of data obtained both from the open literature and from experiments with gas mixtures similar to those in the tanks, we calculated both the flammability in air and the self-flammability of the waste gases in RGS-sampled tanks. The conclusions were that

- $H_2$  dominates flammability, even at relatively low concentrations, and flammability limits for  $H_2$  in the presence of other fuel gases can be readily calculated.
- The minimum waste gas concentration required to reach the LFL in the tank headspace was calculated for the gases retained in nonconvective layers or crusts in the 14 RGS-sampled tanks. The gas fractions, expressed as the volume fraction of gas in the gas/air mixture, ranged from 5.6 vol% (for A-101) to 16 vol% (for U-103).
- Natural inerting (water vapor and waste-gas  $N_2$ ) will not prevent combustion in tank headspaces or in the waste itself in every tank;
- Many wastes did not contain sufficient  $N_2O$  oxidant to support combustion in the absence of air. Of the 15 tanks whose nonconvective or crust layers were sampled by RGS, nine contained gas that was not self-flammable, two (AX-101 and AN-104) contained marginally self-flammable gas, and the remaining four (S-102, SX-106, U-103, and SY-101 crust) contained gas that was probably self-flammable.
- Nitrous oxide does not act as an oxidant until the temperature approaches 1000K. Thus the flammability of waste gas in air near the LFL is not affected by the concentration of nitrous oxide and should not be included in LFL calculation.

When a flammable gas mixture is exposed to an ignition source, the resulting combustion propagation may be either deflagration or detonation. In an open volume, deflagration is possible whenever the gas mixture is between the LFL and the UFL. However, the propagation of deflagration and detonation waves in the pores of a porous medium or a small-diameter pipe depends not only on the gas composition but also on the pore diameter. For assessing Hanford waste, "pore" denotes not only pores between particles but also other, larger cracks and bubble networks. Propagation of detonation in pores, like flame propagation, requires that the pore diameter be greater than a minimum value. The conditions under which detonation can occur (whether in an open space or in pores) are more restrictive than the conditions that permit deflagration.

Deflagration results in subsonic flame propagation away from the ignition site. The peak pressure caused by combustion depends on the completeness of combustion, the resulting adiabatic flame temperature, and the increase (or decrease) in gas moles dictated by the stoichiometry of the combustion reaction. A detonation causes supersonic flame propagation. The peak pressure of a detonation is not uniform throughout an enclosure, as is essentially true for a deflagration, but is localized at the shock wave moving through the gas mixture. For a given gas mixture, a deflagration and a detonation may both release the same amount of energy, but the peak pressure of the detonation shock wave is approximately double that of the deflagration, and the reflection of the shock wave can produce pressures more than twice as high as those in the shock wave itself.

Deflagration and detonation experiments carried out with gas mixtures similar to those retained in the tanks provided information on which an evaluation of RGS-sampled wastes was based, with the following results:

- The measured peak pressures from deflagrations of waste gases were usually more than four times but less than 10 times the initial pressure in all the measurements made with air as the primary oxidant.
- The presence of  $N_2O$  increased the peak pressures of deflagrations, while  $N_2$  significantly reduced the peak pressure.
- Deflagrations are unlikely to propagate within Hanford wastes because retained gas does not appear to take the form of millimeter-diameter pores interconnected in a large network. Creating an ignition source within the waste is also problematic. However, small-scale deflagrations involving fracture bubbles of several cm extent or bubble networks of up to 1 m extent cannot be ruled out.
- $NH_3$  and  $CH_4$  reduced the detonability of  $H_2/N_2O$  mixtures.
- Fuel-rich mixtures (such as the experimentally simulated gases for AN-103 and A-101) remained potentially detonable in the headspace even when diluted 65% by air.
- Detonations will not propagate within Hanford wastes; the same conditions that make deflagration unlikely make detonation implausible.

The final consideration in flammability assessment is the ignition energy required to initiate a sustainable combustion. The minimum ignition energy (MIE) is the energy below which the ignition of a combustible mixture cannot occur and above which ignition occurs. The MIE for a gas varies with the composition of the mixture and is orders of magnitude higher at the LFL than at the slightly fuel-rich conditions at which the MIE is a minimum. Experiments with simulated tank waste gas mixtures showed that 0.04 J sparks were as effective in igniting gas/air mixtures at the LFL as 8 J sparks.

Friction sparks that result from the impact of two surfaces can serve as an ignition source for flammable gas mixtures but cool quickly and can ignite gases only under very favorable conditions. A study showed that hotspot temperatures achieved by normal or glancing blows were less than the temperature required to ignite hydrogen-air mixtures, that the presence of rust could make ignition more likely, and that stainless steel was one of the most sparking-resistant materials available. The results of this work led to using stainless steel for new components that were placed in SY-101 and other similar tanks. In addition, special tools of copper-beryllium and other low-sparking alloys are being used at Hanford to minimize the danger of sparks in hazardous locations.

Flammable gas mixtures can also be ignited by a heated surface that exceeds the auto-ignition temperature. However, for the  $H_2-N_2O-NH_3$  mixtures in Hanford tanks, the auto-ignition temperature would be 457° to 507°C. Procedural protections restrict welding and other activities that might produce such high surface temperatures.

The overall conclusion of this study is that most of the RGS-sampled tanks, and probably most of the tanks at Hanford, contain gas whose composition is such that it can be readily ignited and burned if sufficient gas is released rapidly enough into a tank's headspace. However, it is unlikely that such a deflagration in the headspace will make the transition to a detonation. It is very unlikely that a deflagration could propagate in the retained gas within the waste, and implausible that detonation could be achieved within the waste.

## Reference

Barker SA, WB Barton, DR Bratzel, M Epstein, PA Gauglitz, GD Johnson, SN Maruvada, CE Olson, ML Sauer, SE Slezak, CW Stewart, and J Young. 1999. *Flammable Gas Safety Analysis Data Review*. SNL-000198, Sandia National Laboratories, Albuquerque, New Mexico.

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## 1.0 Introduction

The principal flammable products of Hanford waste gas generation reactions are  $H_2$ ,  $N_2O$ ,  $NH_3$ , and, to a lesser degree,  $CH_4$  and other hydrocarbons. These substances have been observed in many laboratory studies with simulants and in laboratory studies with actual waste samples (Ashby et al. 1992, 1993, 1994; Barefield et al. 1995, 1996; Bryan and Pederson 1993, 1994, 1995; Camaioni et al. 1997; Delegard 1980; Johnson 1997; King et al. 1997; Meisel et al. 1991a, 1991b, 1992, 1993; Person 1996; Pederson and Bryan 1996). These same products are found in the headspaces of actual waste tanks and in gas samples obtained using the RGS (Johnson 1997; Mahoney et al. 1999). All the gases but  $NH_3$  and, to a lesser degree,  $N_2O$  are nearly insoluble in the concentrated, caustic wastes and are found primarily in the gas phase.

The waste gases are often classified by their solubility in the aqueous waste phase and may be divided into low-solubility gases and soluble gases. The low-solubility gases important to flammability are  $H_2$ ,  $N_2O$ , and  $N_2$ .  $CH_4$  and other low-solubility hydrocarbons are also permanent but have negligible effect on solubility. Although  $N_2O$  is about 50 times more soluble than the other low-solubility gases, its solubility is so low that less than 15% of its inventory is dissolved in the liquid. Other low-solubility gases are quantitatively present only as gas.  $NH_3$  is the only soluble gas important to flammability with more than 98% of its inventory present in dissolved form. The relative solubilities of the flammable gases control their forms of release and their effects on headspace flammability. Low-solubility gas releases are dominated by bubble release; soluble gas releases are dominated by direct evaporation from wetted surfaces.  $N_2O$  falls in between but is closer to the other low-solubility gases in behavior. It reaches the headspace predominantly through bubble release, but its headspace concentrations can be significantly affected by evaporation. Thus retained gas compositions can contain lower  $NH_3$  and  $N_2O$  fractions than waste gas measured in the headspace, where evaporation also has an effect.

Thus the gases generated and released by the waste are usually mixtures of flammable gases (primarily  $H_2$  and  $NH_3$ ), a potential oxidant gas ( $N_2O$ ), and one or more inert gases (mainly  $N_2$ ). Whether a flammable condition develops when gas is released depends on the composition and flammability of the gas mixture, the rate at which it is released, the surface area through which it is released, the total volume released, and the rate of mixing and dilution in the headspace.

This report gathers information from several sources to provide an overview of the flammability of waste gases released into the tank headspaces, focusing on the gas properties that must be considered to assess the risk of a deflagration or detonation of the gases released into the tank headspace. Other documents discuss the additional flammability factors, including the release rates (Stewart et al. 1996; Meyer et al. 1997; Barker et al. 1999), the measured ventilation rates (Huckaby et al. 1997a, 1997b, 1998), and the rate at which the waste gases mix with the air in the headspace (Antoniak and Recknagle 1997; Epstein 1995; Epstein and Burelbach 1998).<sup>(a)</sup>

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(a) Antoniak ZI and KP Recknagle, 1995, *Modeling of Post-GRE Spatial and Temporal Hydrogen Concentrations in the Tank AW-101 Dome*, PNLFG:080295; 1996, *Modeling of Post-GRE Spatial and Temporal Hydrogen Concentrations in the Tank AN-105 Dome*, WTS FG95.96; *Modeling of Hydrogen Plume Concentrations in a Single-Shell Tank Dome*, TWSFG96.6; *Modeling of Hydrogen Plume Concentrations in Single and Double-Shell Tank Domes*, TWSFG96.12. PNNL, Richland, Washington.

The following are the specific areas covered in this report:

- Waste gas composition data from headspace and retained gas samples (Section 2)
- Flammability limits applicable to  $H_2/N_2O/NH_3$ /air mixtures and to the retained gas in the absence of air (Section 3)
- Deflagration and detonation in the headspace and in the gas retained in the pores of the waste (Section 4)
- Requirements for and availability of ignition sources (Section 5).

## 2.0 Waste Gas Composition

The waste gas composition measurements that we discuss can be divided into four types based on where the sample was collected:

- the tank headspace
- a waste sample containing trapped gas
- the waste core sampling drill string
- a laboratory gas-generation experiment using waste samples.

Tank headspace sample collection and interpretation are relatively simple and inexpensive. Samples have been collected on a majority of the tanks. By contrast, direct sampling of the gas retained by the waste involves a specialized waste sampling system, the retained gas sampler (RGS), and has been performed successfully on only 14 tanks. Gas samples collected from the drill string of the waste core sampling system have been analyzed for 12 tanks. Like the RGS samples, they provide information on the composition of trapped gas but have large uncertainties due to sampling conditions. Laboratory gas generation tests have been conducted on wastes from six tanks and are the only source of information on the effect of temperature.

Each method has its own limitations, as discussed in this section. Data for retained gas composition and headspace gas composition generally do not agree well. The reason for this is not understood; however, the RGS data are believed to best define the retained gas composition, though there is high uncertainty in the retained ammonia, and thus to best define the composition of the gas released to the headspace during GREs (Barker et al. 1999). The headspace gas composition best defines the flammability for steady-state conditions.

Waste gas released into the tank headspaces is diluted by the air present, so comparisons of headspace waste gas concentrations with results from RGS, drill string, and laboratory tests are typically done by comparing concentration ratios (e.g.,  $H_2/N_2O$ ,  $NH_3/N_2O$ ,  $H_2/N_2$ , etc.). Ratios involving  $N_2$  are not useful because  $N_2$  in the tank headspaces (and some of the drill string samples) is overwhelmingly dominated by  $N_2$  from air. The waste  $N_2$  contribution to the headspace concentration is not measurable. The  $H_2/N_2O$  ratio is particularly useful because it is the dominant fuel/oxidant ratio under airless conditions in gas pockets in the waste, and it tends to be indicative of different waste types. The  $H_2/NH_3$  ratio tends not to be useful as a means of comparison, primarily because the ammonia concentration in trapped gas bubbles is essentially in equilibrium with the surrounding waste, while the ammonia concentration in the headspace is controlled by the evaporation rate. However, the headspace  $H_2/NH_3$  ratio does indicate to what extent  $H_2$  dominates the flammability of headspace gas.

Comparisons of the four types of waste gas composition measurements and their gas concentration ratios are presented and discussed in the following sections. Headspace gas compositions are discussed in Section 2.1, retained gas compositions are discussed in Section 2.2, core sample drill string gas compositions in Section 2.3, and waste sample gas compositions derived from laboratory tests in Section 2.4.

## 2.1 Headspace Gas Composition

The vast majority of waste gas composition data have come from the tank headspaces or ventilation systems connected to the headspaces. The concentration of a waste gas in a tank headspace at any time is a function of the concentration of the gas at some initial time and the rates of gas release by the waste and ventilation since the initial time. To the extent that the gas release and ventilation rates are constant, the concentrations of waste gases will approach constant (steady state) values.

One limitation of headspace measurements is that no information can be obtained about the release of  $N_2$  by the waste. This is because  $N_2$  is the principal constituent of air and the change due to released quantities is not measurable; that is, it is a problem of measuring very small changes in a large value. In the context of this report, the  $N_2$  present in waste gas releases is primarily of interest because it serves as a diluent.

Another limitation of headspace composition measurements is that headspace gas and vapor concentrations are inherently based on the rates of gas release from the waste surface. They may not be representative of the retained gases within the waste and of the headspace composition after large releases. Gases that are soluble in the aqueous waste are transported to the waste surface via diffusion and convection in the aqueous phase, where they may be released by evaporation into the headspace. Less soluble gases are not efficiently transported by this mechanism; they tend to build up in the retained gas phase and appear when the gas is released. Evidence of differences between headspace and retained gas compositions is discussed in Section 2.5.

A further limitation of headspace composition measurements is that headspace gas and vapor concentrations fluctuate with time. Fluctuations are due principally to changes in temperatures and ventilation rates. Most of the single-shelled tanks (SSTs) are passively ventilated, and the exchange of air between their headspaces and the atmosphere can be affected by changing meteorological conditions. In actively ventilated tanks, which are ventilated by exhaust fans, fluctuations result primarily from changes in fan operation.

Available data suggest that fluctuations of more than about 40% over a period of months are possible but not common in the passively ventilated tanks. The interpretation of headspace composition data in passively ventilated tanks is further complicated by the fact that cascade line connections and ventilation system piping between tanks can allow significant exchange between the headspaces of adjacent tanks. This leads to potential errors in assuming the gases in a tank's headspace are from the waste in that tank. Exchanges between tanks, evidently via a cascade line, have been observed during tracer ventilation studies. The airflow through the cascade line was also observed to change direction (Huckaby et al. 1998). These limitations notwithstanding, the headspace composition data indicate that levels of flammable gases, including  $CH_4$  and  $NH_3$ , have been lower than their lower flammability limits (LFLs) in air.

Headspace flammable gas composition data have been obtained by two sampling and analytical methods as well as various gas monitoring methods. Gas "grab" samples have been collected routinely using the standard hydrogen monitoring systems (SHMS) mounted on selected tanks (McCain 1999). Gas samples have also been collected from 110 SSTs using

SUMMA canisters and sorbent traps (PNNL 1999). Various analytical methods have been applied to measure low-solubility gases,  $\text{NH}_3$ , and organic vapors in these samples. Because the laboratory analyses are consistent with and representative of the higher-quality monitoring data, only the laboratory analyses are presented and discussed further in this report.

Monitoring methods range from real-time industrial hygiene “sniffs” to field-installed, data-logged automated analytical systems. Industrial hygiene sniffs are routinely performed to ensure safe working conditions. They are conducted with portable organic vapor monitors (OVMs), organic vapor analyzers (OVAs), combustible gas monitors (CGMs), and colorimetric indicator tubes. The suite of automated instrumentation include Whittaker cells for monitoring  $\text{H}_2$  concentrations, photoacoustic infrared spectrometers for measuring  $\text{NH}_3$ , a Fourier transform infrared (FTIR) spectrometer that also measures ammonia, and gas chromatographs with thermal conductivity detectors (GC-TCD) for monitoring  $\text{H}_2$ ,  $\text{N}_2\text{O}$ , and  $\text{CH}_4$ . The monitoring data constitute a large body of information but are of varying quality. Only the laboratory analyses are presented in this report.

### **2.1.1 SHMS Cabinet Grab Sample Methods**

Gas grab samples are routinely collected manually from the SHMS cabinets for laboratory analysis. Grab samples are also collected automatically when a high-level alarm is activated by a monitoring instrument. A description of the sampling system and the analytical method is given by McCain (1999). The SHMS grab samples have been analyzed for permanent gases by mass spectrometry at the PNNL Radiochemical Processing Laboratory (RPL).

Table 2.1 lists the 42 double-shelled tanks (DSTs) and SSTs for which SHMS grab samples have been collected, the number of samples analyzed, the average and maximum  $\text{H}_2$  concentration reported, and the average ratio of  $\text{H}_2$  to  $\text{N}_2\text{O}$  in the samples (McCain 1999). The  $\text{H}_2$  averages include the reported detection limits when no  $\text{H}_2$  was detected in the sample. The average  $\text{H}_2$  to  $\text{N}_2\text{O}$  ratio includes only samples for which both  $\text{H}_2$  and  $\text{N}_2\text{O}$  concentrations were above detection limits. “U” is entered when no samples from a given tank met that criterion.

Boldface type indicates that each maximum reported  $\text{H}_2$  concentrations in AN-103, AN-104, AN-105, AW-101, and SY-103 is associated with a GRE and was collected automatically by a high-level alarm. Though these GRE-related concentrations are higher than the corresponding averages, they are probably lower than those measured by on-line instruments for the given GRE. Aside from the results associated with GREs, the higher  $\text{H}_2$  concentrations tend to be associated with passively ventilated SSTs in which breathing rates are highly variable and usually at least an order of magnitude less than in actively ventilated tanks. All of these measured hydrogen concentrations are well below the LFL of hydrogen at 40,000 ppmv in air.

### **2.1.2 SUMMA Canister and Sorbent Trap Sample Methods**

Concentrations of permanent gases, inorganic vapors, and organic vapors have been measured in 110 SSTs using SUMMA canister and sorbent trap sampling methods and laboratory analyses. SUMMA canisters are stainless steel vessels with electrochemically passivated internal surfaces that improve the recovery of many organic vapors. SUMMA

**Table 2.1. SHMS Cabinet Grab Sample Data**

Tank	Type	Number of H <sub>2</sub> Samples	Average H <sub>2</sub> (ppmv)	Average H <sub>2</sub> /N <sub>2</sub> O	Maximum H <sub>2</sub> (ppmv)
A-101	SST, P	16	801	4.8	1,540
AN-101	DST, A	15	10	U	10
AN-103	DST, A	62	44	11	<b>800</b>
AN-104	DST, A	57	27	6.2	<b>154</b>
AN-105	DST, A	35	476	5.3	<b>10,700</b>
AN-107	DST, A	20	23	0.3	49
AW-101	DST, A	39	426	43	<b>2,980</b>
AX-101	SST, P	19	48	9.6	100
AX-103	SST, P	9	26	1.1	36
AY-102	DST, A	32	33	U	73
AZ-101	DST, A	6	37	U	65
AZ-102	DST, A	4	98	U	190
BY-103	SST, P	18	54	3.3	230
BY-105	SST, P	4	35	0.8	69
BY-106	SST, P	18	143	1.9	1,110
BY-109	SST, P	18	36	10	154
C-106	SST, A	2	10	U	10
S-101	SST, P	6	332	1.6	430
S-102	SST, P	12	568	1.3	760
S-106	SST, P	5	19	U	30
S-107	SST, P	6	137	2.2	180
S-108	SST, P	2	12	U	12
S-109	SST, P	4	44	U	64
S-111	SST, P	21	81	7.3	330
S-112	SST, P	22	25	6.9	43
SX-101	SST, A	24	9	3.6	33
SX-102	SST, A	17	17	3.5	42
SX-103	SST, A	24	27	4.2	66
SX-104	SST, A	28	8	U	28
SX-105	SST, A	30	22	4.3	98
SX-106	SST, A	25	39	3.0	160
SX-109	SST, A	18	19	4.0	69
SY-101	DST, A	9	14	0.9	22
SY-102	DST, A	15	10	U	10
SY-103	DST, A	56	229	2.0	<b>1,810</b>
T-110	SST, P	21	9	U	21
U-102	SST, P	4	335	0.4	540
U-103	SST, P	21	575	0.7	840
U-105	SST, P	18	611	0.4	1,550
U-107	SST, P	24	390	0.8	690
U-108	SST, P	29	379	1.0	1,060
U-109	SST, P	20	386	0.8	630

DST–double shelled tank; SST–single shelled tank.  
A–actively ventilated; P–passively ventilated.  
U–less than detection limit.

canisters used at Hanford were cleaned and evacuated at a laboratory, then filled with air from the tank headspace air via a purged sampling probe. Sorbent traps are pencil-sized tubes packed with sorbent media that remove specific vapors from an air sample passed through them. Sorbent traps were used for ammonia and organic vapors. Generally, tanks have been sampled on a single date from the central portion of the headspace, but some SSTs have been resampled periodically to examine temporal effects (Huckaby et al. 1997a) and from multiple locations to examine headspace homogeneity (Huckaby et al. 1997b). Sampling and analysis methods are described by Huckaby et al. (1996).

Table 2.2 lists the average NH<sub>3</sub>, H<sub>2</sub>, CH<sub>4</sub>, and N<sub>2</sub>O concentrations measured in each of the 110 SST headspaces that have been sampled. Also given in Table 2.2 are the H<sub>2</sub>/N<sub>2</sub>O ratio, the highest reported total non-methane hydrocarbon (TNMHC) concentration, and the calculated maximum percentage of the LFL in air for each tank.

**Table 2.2.** Headspace Flammable Species Average Concentrations

	Type	NH <sub>3</sub> (ppmv)	CH <sub>4</sub> (ppmv)	H <sub>2</sub> (ppmv)	N <sub>2</sub> O (ppmv)	Ratio <sup>(a)</sup> of H <sub>2</sub> /N <sub>2</sub> O	TNMHC <sup>(b)</sup> (mg/m <sup>3</sup> )	Max % of LFL <sup>(c)</sup>
		LFL in air 150,000	LFL in air 50,000	LFL in air 40,000	NA <sup>(d)</sup>	NA	LFL in air 42,000 <sup>(e)</sup>	NA
241-A-101	SST, P	754	< 12	758	218	3.5	26	2.5
241-A-102	SST, P	257	< 4	297	227	1.3	5.0	0.9
241-A-103	SST, P	264	< 4	474	222	2.1	7.8	1.4
241-A-106	SST, P	278	< 50	120	200	0.6	4.8	0.7
241-AX-101	SST, P	42	< 12	103	< 13	> 8	3.3	0.3
241-AX-102	SST, P	34	< 12	< 98	50	< 2	12	0.3
241-AX-103	SST, P	42	< 12	< 98	24	< 4	0.5	0.3
241-AX-104	SST, P	< 14	< 110	< 100	< 100	NA	1.1	0.6
241-B-102	SST, P	5	< 25	< 17	64	< 0.3	4.2	0.1
241-B-103	SST, P	9	< 54	< 44	66	< 0.7	17	0.4
241-B-105	SST, P	6	< 25	< 17	27	< 0.6	3.8	0.1
241-B-107	SST, P	21	< 25	< 17	< 17	NA	3.6	0.1
241-B-202	SST, P	3	< 25	< 17	< 17	NA	0.9	0.1
241-BX-102	SST, P	3	< 25	< 17	< 17	NA	7.9	0.1
241-BX-103	SST, P	91	< 25	154	80	1.9	55	0.6
241-BX-104	SST, P	213	< 31	91	132	0.69	130	0.8
241-BX-105	SST, P	147	< 25	49	117	0.42	4.4	0.3
241-BX-106	SST, P	47	< 25	< 17	44	< 0.4	2.0	0.1
241-BX-107	SST, P	83	< 4	11	12	0.92	2.4	0.1
241-BX-110	SST, P	63	< 25	< 17	< 17	NA	3.8	0.2
241-BX-111	SST, P	63	< 25	< 17	< 17	NA	11	0.2
241-BY-101	SST, P	60	< 25	29	57	0.51	57	0.3
241-BY-102	SST, P	175	< 4	34	18	1.9	20	0.3
241-BY-103	SST, P	30	< 61	21	33	0.64	13	0.4
241-BY-104	SST, P	266	8	250	253	1.0	207	1.4
241-BY-105	SST, P	50	4	85	86	1.0	18	0.3
241-BY-106	SST, P	82	3.6	55	82	0.67	13	0.3

**Table 2.2. Headspace Flammable Species Average Concentrations**

	Type	NH <sub>3</sub> (ppmv)	CH <sub>4</sub> (ppmv)	H <sub>2</sub> (ppmv)	N <sub>2</sub> O (ppmv)	Ratio <sup>(a)</sup> of H <sub>2</sub> /N <sub>2</sub> O	TNMHC <sup>(b)</sup> (mg/m <sup>3</sup> )	Max % of LFL <sup>(c)</sup>
241-BY-107	SST, P	972	NA	692	712	1.0	173	1.9
241-BY-108	SST, P	881	13	431	589	0.73	594	2.9
241-BY-109	SST, P	45	NA	NA	NA	NA	30	0.1
241-BY-110	SST, P	401	< 61	< 160	103	< 1.6	50	1.0
241-BY-111	SST, P	59	< 61	67	99	0.68	10	0.6
241-BY-112	SST, P	63	< 61	< 94	40	< 2.4	14	0.4
241-C-101	SST, P	99	12	436	643	0.68	256	1.4
241-C-102	SST, P	189	5	146	143	1.0	313	0.9
241-C-103	SST, P	328	16	776	717	1.1	2139	3.5
241-C-104	SST, A	44		70	64	1.1	27	0.3
241-C-105	SST, A	2		22	5	4.4	7.5	0.1
241-C-106	SST, A	3		10	4	2.5	1.3	0.03
241-C-107	SST, P	64	3.4	455	107	4.3	10	3.3
241-C-108	SST, P	3	0.3	15	180	0.08	1.4	0.04
241-C-109	SST, P	10	0.93	125	369	0.34	1.8	0.3
241-C-110	SST, P	124	2	12	15	0.80	28	0.5
241-C-111	SST, P	3	0.5	14	69	0.20	1.2	0.0
241-C-112	SST, P	23	1	204	544	0.38	6.3	0.5
241-C-201	SST, P	< 2	< 25	< 17	< 17	NA	6.3	0.1
241-C-202	SST, P	< 1	< 25	< 17	< 17	NA	2.5	0.1
241-C-204	SST, P	1	< 25	< 17	< 17	NA	218	0.5
241-S-101	SST, P	773	< 25	442	271	1.6	15	1.7
241-S-102	SST, P	600	6	742	701	1.1	68	3.7
241-S-103	SST, P	150	< 50	79	136	0.58	13	0.4
241-S-105	SST, P	37	< 4	21	8	2.6	2.7	0.1
241-S-106	SST, P	36	< 25	< 17	< 17	NA	9.2	0.1
241-S-107	SST, P	183	< 25	362	91	4.0	7.1	1.1
241-S-108	SST, P	26	< 4	22	3	7.3	2.6	0.1
241-S-109	SST, P	45	< 25	< 17	< 17	NA	3.7	0.1
241-S-110	SST, P	148	< 4	135	40	3.4	3.8	0.5
241-S-111	SST, P	122	< 23	391	48	8.1	2.0	1.1
241-S-112	SST, P	89	< 25	< 25	< 25	NA	8.2	0.2
241-SX-101	SST, A	4	< 25	< 25	< 25	NA	1.0	0.1
241-SX-102	SST, A	16	< 25	< 25	< 25	NA	5.6	0.2
241-SX-103	SST, A	77	< 23	< 23	< 23	NA	1.0	0.2
241-SX-104	SST, A	25	< 25	< 25	< 25	NA	1.6	0.2
241-SX-105	SST, A	28	< 25	< 25	< 25	NA	3.0	0.2
241-SX-106	SST, A	179	< 12	< 98	14	< 7	2.3	0.4
241-SX-107	SST, A	< 14	< 50	< 50	< 50	NA	0.7	0.3
241-SX-108	SST, A	< 14	< 50	< 50	< 50	NA	1.1	0.3
241-SX-109	SST, A	17	< 25	< 25	< 25	NA	1.2	0.2
241-SX-110	SST, A	< 14	< 50	< 50	< 50	NA	1.0	0.3
241-SX-111	SST, A	< 14	< 50	< 50	< 50	NA	1.1	0.3
241-SX-112	SST, A	< 14	< 50	< 50	< 50	NA	1.1	0.3
241-SX-114	SST, A	< 14	< 50	< 50	< 50	NA	0.9	0.3

**Table 2.2. Headspace Flammable Species Average Concentrations**

	Type	NH <sub>3</sub> (ppmv)	CH <sub>4</sub> (ppmv)	H <sub>2</sub> (ppmv)	N <sub>2</sub> O (ppmv)	Ratio <sup>(a)</sup> of H <sub>2</sub> /N <sub>2</sub> O	TNMHC <sup>(b)</sup> (mg/m <sup>3</sup> )	Max % of LFL <sup>(c)</sup>
241-T-101	SST, P	22	< 50	52	< 50	> 1.0	0.2	0.3
241-T-104	SST, P	105	< 4	12	8	1.5	1.9	0.1
241-T-107	SST, P	125	< 61	< 94	42	< 2.2	3.9	0.5
241-T-110	SST, P	108	< 25	< 25	< 25	NA	1.1	0.2
241-T-111	SST, P	226	< 61	< 94	< 13	NA	24	0.6
241-TX-101	SST, P	< 14	< 50	69	< 50	> 1.4	1.1	0.3
241-TX-102	SST, P	170	< 50	< 50	55	< 0.9	10	0.4
241-TX-103	SST, P	108	< 50	< 50	87	< 0.6	5.2	0.3
241-TX-104	SST, P	29	< 50	< 50	< 50	NA	1.5	0.3
241-TX-105	SST, P	20	< 61	< 99	13	< 8	5.2	0.4
241-TX-106	SST, P	255	< 150	< 50	< 50	NA	5.6	0.6
241-TX-108	SST, P	25	< 50	< 50	< 50	NA	3.8	0.3
241-TX-110	SST, P	985	< 50	160	130	1.2	30	1.2
241-TX-111	SST, P	612	< 25	109	177	0.62	16	0.8
241-TX-112	SST, P	113	< 50	< 50	< 50	NA	7.0	0.3
241-TX-113	SST, P	22	< 50	< 50	< 50	NA	7.7	0.3
241-TX-114	SST, P	163	< 50	< 50	< 50	NA	11	0.4
241-TX-115	SST, P	< 14	< 50	< 50	< 50	NA	3.6	0.3
241-TX-116	SST, P	14	< 50	< 50	< 50	NA	15	0.3
241-TX-117	SST, P	< 14	< 50	< 50	< 50	NA	4.0	0.3
241-TX-118	SST, P	38	3	78	18	4.3	12	0.4
241-TY-101	SST, P	17	< 12	< 93	98	< 1.0	22	0.3
241-TY-102	SST, P	4	< 4	< 3	67	< 0.04	1.0	0.0
241-TY-103	SST, P	44	< 21	63	190	0.33	95	0.4
241-TY-104	SST, P	59	< 23	< 49	98	< 0.5	20	0.2
241-TY-105	SST, P	< 14	< 50	< 50	51	< 1.0	35	0.3
241-U-102	SST, P	658	< 50	1100	1100	1.0	10	3.3
241-U-103	SST, P	730	< 61	555	878	0.63	11	2.0
241-U-104	SST, P	< 1	< 25	< 17	86	< 0.2	0.6	0.1
241-U-105	SST, P	325	< 23	< 49	154	< 0.3	4.8	0.4
241-U-106	SST, P	931	< 61	210	559	0.38	17	1.3
241-U-107	SST, P	453	< 12	500	701	0.71	16	1.6
241-U-108	SST, P	692	< 25	522	612	0.85	12	1.9
241-U-109	SST, P	577	< 25	748	868	0.86	12	2.3
241-U-111	SST, P	676	< 12	247	327	0.76	9.3	1.1
241-U-112	SST, P	305	< 25	142	254	0.56	6.8	0.9
241-U-203	SST, P	1	< 25	< 25	< 25	NA	11	0.1
241-U-204	SST, P	.1	< 25	< 25	< 25	NA	7.2	0.1

A – actively ventilated. P – passively ventilated.

(a) Calculated from average hydrogen and nitrous oxide concentrations given in columns 5 and 6.

(b) TNMHC = total non-methane hydrocarbons. Value is highest of average EPA Task Order 12 analysis for TNMHC, and the average summation of organic species based on gas chromatograph/mass spectral analyses.

(c) %LFL = Maximum in Tank Characterization Database (PNNL 2000) under vapor flammability results.

(d) NA = not available or not applicable.

(e) LFL of TNMHC in air was based on LFL values for semivolatile alkanes.

Compared with their LFLs in air, the measured headspace concentrations of all the flammable species are small. Of all the headspace sampling events conducted, the highest calculated flammability was observed for Tank S-102, for which the fuel content reached only about 3.7% of the LFL. The conclusion to be drawn is that, under normal conditions, the SST headspaces contain far too little fuel to be flammable.

## 2.2 Retained Gas Composition

Direct sampling of the gas retained by the waste has been performed successfully on 14 tanks using the RGS, which is a modified version of the universal core sampler specifically designed to sample gas retained in tank waste. The results of analysis of RGS samples are used to calculate the gas volume fraction and the partition of gas species between the liquid and gas phases under in situ conditions. The major species generally measured are H<sub>2</sub>, NH<sub>3</sub>, N<sub>2</sub>, N<sub>2</sub>O, O<sub>2</sub>, argon, CH<sub>4</sub>, and other hydrocarbons (a lumped category). The quantity of in situ ammonia is measured with significant uncertainty because of the limitations of the analytical process. Details on the core sampler and modifications and operational constraints of the RGS are contained in Webb (1994).

The analysis of RGS samples is carried out by extruding hermetically sealed core samples into an evacuated sealed container. The gases released by the extruded sample are pumped into collection canisters whose pressure, temperature, and composition are measured, allowing calculation of the amount and composition of the gas in the sample. More detailed descriptions of the experimental procedure may be found in Shekarriz et al. (1997) and Mahoney et al. (1997, 1999). Comprehensive studies were performed to quantify the uncertainty in measuring retained gases using RGS (Cannon and Knight 1995; Cannon 1996).

Gas concentrations are subject to small uncertainties (less than 15%) for low-solubility species such as H<sub>2</sub>, N<sub>2</sub>O, N<sub>2</sub>, and CH<sub>4</sub> and higher uncertainties (at least 50%) for NH<sub>3</sub>. Once the concentration of each species in the sample is determined, the in situ partition of the species between the gas and liquid phases is calculated using the Schumpe gas solubility model. The overall uncertainty in the reported composition of the retained gas (the gas phase) is derived from the uncertainties in both the measured species concentrations and the solubility model.

### 2.2.1 Low-Solubility Species

Tables 2.3, 2.4, and 2.5 show the mole fractions of hydrogen, nitrogen, and nitrous oxide, respectively, in the tanks that have been sampled with the RGS (Mahoney et al. 1999). The tables are arranged by row in chronological order of sampling and are subdivided into columns according to the types of layers from which samples were taken. The average mole fraction for each layer is given, together with the number of samples taken in each layer (in parentheses). Within each cell of the table, the mole percent for a lower-bound solubility is given first with its measurement uncertainty followed by the mole percent for an upper-bound solubility. In some cases, the same mole percent was calculated for both ends of the solubility range, so a single number is presented.

**Table 2.3.** Hydrogen Mole Fractions in Retained Gas, from RGS Sampling

Tank	Average in a layer located in or immediately under surface solids (mol%)	Average in a convective or mixed layer (mol%)	Average in a settled solids layer (mol%)
A-101	70±7.3–72±7.0 (4)	14±4.3–15±4.6 (3)	
AN-103	62±6.4–63±6.4 (1)	19±12 (2)	61±7.7–62±7.7 (2)
AN-104		24±13–26±15 (1)	45±6.9–47±7.0 (5)
AN-105		25±12 (1)	59±5.4–62±5.4 (6)
AW-101		26±9.4–25±9.2 (1)	32±3.2–33±3.2 (5)
AX-101	60±5.5–64±5.4 (1)		
BY-109			50±5.5–51±5.6 (3)
S-102			33±3.0–34±3.0 (1)
S-106			63±5.7–64±5.9 (4)
S-111		6.4±3.4–6.1±3.3 (1)	66±1–67±11 (4)
SX-106		21±4.8–18±4.4 (2)	51±4.7–53±4.5 (5)
SY-101	34±4.4–40±4.5 (5)	26±8.1–31±9.5 (7)	
U-103			23±1.4 (4)
U-109			25±3.0–26±3.1 (4)

**Table 2.4.** Nitrogen Mole Fractions in Retained Gas, from RGS Sampling

Tank	Average in a layer located in or immediately under surface solids (mol%)	Average in a convective or mixed layer (mol%)	Average in a settled solids layer (mol%)
A-101	19±4.9 (4)	60±22–71±27 (3)	
AN-103	29±3.2 (1)	68±61–75±68 (2)	33±4.3 (2)
AN-104		55±48–63±57 (1)	29±4.8–31±5.0 (5)
AN-105		57±41–64±47 (1)	24±4.0 (6)
AW-101		67±32–69±33 (1)	55±6.2–57±6.3 (5)
AX-101	16±2.6–18±2.7 (1)		
BY-109			28±5.0–29±5.1 (3)
S-102			32±4.3–33±4.4 (4)
S-106			25±3.7–26±3.8 (4)
S-111		90±68–91±69 (1)	21±5.6–22±5.7 (4)
SX-106		61±20–76±25 (2)	20±4.0–21±4.0 (5)
SY-101	27±4.5–32±4.9 (5)	40±14–48±16 (7)	
U-103			36±2.3–37±2.3 (4)
U-109			46±7.7–47±8.0 (4)

**Table 2.5.** Nitrous Oxide Mole Fractions in Retained Gas, from RGS Sampling

Tank	Average in a layer located in or immediately under surface solids (mol%)	Average in a convective or mixed layer (mol%)	Average in a settled solids layer (mol%)
A-101	5.7±0.6–5.5±0.5 (4)	15±5.4–6.0±2.2 (3)	
AN-103	6.9±0.7–6.3±0.6 (1)	8.3±5.3–2.6±1.7 (2)	4.2±0.6–3.6±0.5 (2)
AN-104		15±8.4–5.2±3.1 (1)	23±3.8–20±3.3 (5)
AN-105		13±6.6–4.6±2.4 (1)	15±1.5–12±1.2 (6)
AW-101		2.3±0.8–0.8±0.3 (1)	7.5±0.8–5.1±0.6 (5)
AX-101	11±1.1 (1)		
BY-109			18±2.5–17±2.3 (3)
S-102			33±3.3–32±3.0 (4)
S-106			11±1.0–9.0±0.9 (4)
S-111		2.2±1.3–0.9±0.6 (1)	11±1.8–10±1.6 (4)
SX-106		15±4.2–2.8±0.8 (2)	24±2.7 (5)
SY-101	19±2.5–22±2.4 (5)	24±7.4–16±4.9 (7)	
U-103			40±2.4–38±2.3 (4)
U-109			27±3.6–25±3.4 (4)

### 2.2.2 Ammonia

Ammonia is a highly soluble species. Little of the sample  $\text{NH}_3$  is present in the gas extracted from RGS samples by the analytical methods that have been used; most of it remains dissolved in the sample liquid. To determine the residual dissolved  $\text{NH}_3$ , it is necessary to either measure it directly or find a physical relationship by which the residual  $\text{NH}_3$  can be calculated from the extracted  $\text{NH}_3$ . Both alternatives are technically problematic, and the actual relationship between the extracted  $\text{NH}_3$  vapor and the dissolved residual  $\text{NH}_3$  is difficult to determine. Therefore, the uncertainty in ammonia concentrations is estimated as at least 50%. See Mahoney et al. (2000) for a detailed discussion of ammonia extraction procedures, analysis methods, and the resulting uncertainties. The  $\text{NH}_3$  mole fractions in the retained gas are given in Table 2.6.

### 2.3 Drill String Samples

Drill string samples are grab samples taken from the gas volume in the drill string during tank waste core drilling operations. The primary purpose of drill string samples is to ensure safe working conditions during core drilling. Under certain circumstances, these samples also provide data on gases retained in the tank waste. Siciliano (1998) provides a detailed description of the process of extracting data relevant to characterizing retained gases from this type of sample. These data provide approximate retained gas information on tanks not sampled with the more refined RGS method.

**Table 2.6. Ammonia Mole Fractions in Retained Gas, from Recent RGS Sampling**

Tank	Average in a layer located in or immediately under surface solids (mol%)	Average in a convective or mixed layer (mol%)	Average in a settled solids layer (mol%)
A-101	4.8±1.8 – 1.8±0.7 (4)	7.8±4.0 – 3.0±1.5 (3)	
AN-103	1.8±0.8 – 0.6±0.3 (1)	1.2±0.9 – 0.5±0.4 (2)	0.9±0.4 – 0.3±0.1 (2)
AN-104		1.0±0.7 – 0.4±0.3 (1)	1.4±0.7 – 0.5±0.3 (5)
AN-105		0.8±0.5 – 0.4±0.2 (1)	0.6±0.3 – 0.3±0.1 (6)
AW-101		0.6±0.3 – 0.2±0.1 (1)	0.9±0.4 – 0.3±0.2 (5)
AX-101	9.1±2.1 – 3.5±0.8 (1)		
BY-109			0.3±0.1 – 0.1±0.06 (3)
S-102			1.0±0.3 – 0.4±0.1 (4)
S-106			0.4±0.2 – 0.2±0.1 (4)
S-111		0.1±0.1 (1)	1.0±0.3 – 0.5±0.2 (4)
SX-106		0.6±0.1 – 0.2±0.05 (2)	4.4±0.9 – 1.6±0.4 (5)
SY-101	19±4.7 – 4.9±1.2 (5)	7.1±2.2 – 2.3±0.7 (7)	
U-103			0.6±0.3 – 0.3±0.1 (4)
U-109			0.9±0.3 – 0.4±0.2 (4)

### 2.3.1 Sample Collection

The drill string is the passageway through which core samplers are lowered into the waste and cores are retrieved. Vapors or gases that are released from the waste during the drilling are contained in the headspace of the drill string. Drill string samples potentially contain gas originating in more than one core sample or layer so do not provide a definite basis for making distinctions between the gas composition in different layers.

During the tank waste coring operation, hazardous gas monitoring is used to verify that hazardous gases in the drill string are below specified safety limits. When the limits are exceeded, a grab sample may be collected for later laboratory analysis. In addition, the drill string headspace may be purged with argon or N<sub>2</sub> or air may be admitted to the drill string. Thus, subsequent grab samples can contain residual purge gas or air, which complicates the interpretation of results. The sample containers sometimes also contain residual helium from their preparation.

### 2.3.2 Sample Analysis

The samples were analyzed using mass spectrometry at the RPL. Because oxygen, argon, and helium are not produced in the waste, the measured amounts of these gases were assumed to be residuals from either the sample container preparation or drill string purging and were used to correct for air/purge gas contamination. Siciliano (1998) provides the details of data correction.

### 2.3.3 Results

Table 2.7 lists the 16 drill string samples collected and the concentrations of the primary gases in each sample. Ammonia is not listed because, as in the case of SHMS grab samples, the sampling methods and devices did not allow this species to be properly measured. Table 2.7 also lists the calculated ratios of H<sub>2</sub> to N<sub>2</sub> and H<sub>2</sub> to N<sub>2</sub>O.

**Table 2.7.** Adjusted Gas Analysis Data from Drill String Samples

Tank	H <sub>2</sub> (mol%)	CH <sub>4</sub> (mol%)	N <sub>2</sub> (mol%)	N <sub>2</sub> O (mol%)	Other (mol%)	H <sub>2</sub> /N <sub>2</sub> O Ratio	H <sub>2</sub> /N <sub>2</sub> Ratio
AN-103	47.5	0.203	47.1	5.15	0.33	9.2	1.01
AN-104	33.3	0.897	53.9	11.2	0.56	3.0	0.62
AW-101	25.4		72.6	2.03		12.5	0.35
B-203	3.10	0.0020	96.7		0.28		0.03
BY-109	32.4	0.57	56.9	9.65	1.1	3.4	0.57
BY-110	35.2	1.22	39.0 <sup>(a)</sup>	21.8	2.9	1.6	0.90
S-102	25.8	0.201	58.7 <sup>(a)</sup>	15.4		1.7	0.44
S-102	36.7		60.5 <sup>(a)</sup>	2.98		12.3	0.61
S-106	41.7		47.7	10.4	0.24	4.0	0.87
S-106	64.9	0.176	23.6	10.3	0.38	6.3	2.75
S-107	19	0.0308	75.9	2.52	1.45	7.6	0.25
SX-106	2.31	0.015	96.6	0.924	0.12	2.5	0.02
SY-102	2.42	0.015	96.9	0.307	0.23	7.9	0.02
U-109	41.0	0.171	50.7 <sup>(a)</sup>	7.98	1.74	5.1	0.81
U-109	23.4	0.302	56 <sup>(a)</sup>	20	0.26	1.2	0.42
U-109	23.2	0.392	61.1 <sup>(a)</sup>	15.1		1.5	0.38

(a) Nitrogen purged.

### 2.4 Waste Sample Gas Composition Derived from Laboratory Tests

The rate and composition of waste gas generation has been measured in the laboratory for six tanks under different combinations of temperature and irradiation conditions. This composition measurement method is the only one that allows us to predict gas composition change with temperature and irradiation.

The laboratory gas generation tests involved subjecting tank waste core samples under helium cover gas to different temperatures and levels of radiation (supplied by a <sup>137</sup>Cs gamma source). Runs typically lasted several hundred hours. The stainless-steel reaction vessels were connected to a stainless-steel gas manifold outside the hot cell. Gas samples were taken from the manifold several times during the run and analyzed by mass spectrometry at RPL to determine composition. Argon was measured to determine the correction for nitrogen that leaked in from air. Ammonia was usually detected in these samples but is not included here because a total NH<sub>3</sub> generation rate could not be measured accurately with the gas manifold system. Table 2.8 is a summary of generation rates and concentrations from the experiments (King and Bryan 1999).

**Table 2.8.** Concentrations and Generation Rates of Gases Measured from Waste Samples

Tank	H <sub>2</sub>		N <sub>2</sub> O		N <sub>2</sub>		CH <sub>4</sub>	
	mol/day <sup>(a)</sup>	% of total						
A-101	1.13	45.0	0.73	29.1	0.59	23.5	0.06	2.4
AW-101	11.6	81.2	1.6	11.2	1.0	7.0	0.09	0.6
S-102	1.0	61.8	0.4	24.7	0.2	12.4	0.019	1.2
S-106	0.19	39.8	0.2	41.9	0.08	16.8	0.007	1.5
SY-103	10.7	78.2	0.9	6.6	2.0	14.6	0.08	0.6
U-103	0.48	26.7	0.65	36.1	0.60	33.3	0.07	3.9

(a) Based on tank waste mass, temperature, and dose rate.

## 2.5 Summary of Gas Compositions

As shown in Tables 2.3 and 2.7, the retained gas in the sampled tanks ranged from less than 20 mol% to over 70 mol% H<sub>2</sub>. The results from laboratory gas generation tests also showed substantial variation in the H<sub>2</sub> mole fraction. RGS samples indicated that the H<sub>2</sub> mole fractions in liquid layers were lower than those in settled solids layers, even when the higher uncertainty associated with measuring smaller amounts of gas and proportionally larger contributions from air contamination in the liquid layers is taken into account.

The N<sub>2</sub> and N<sub>2</sub>O mole fractions (Tables 2.4, 2.5, and 2.7) also varied considerably from tank to tank. The highest N<sub>2</sub> values were often found in RGS samples from liquid layers with high uncertainties. In nonconvective and crust layers, the N<sub>2</sub> mole fraction in most cases lay between one and three times the N<sub>2</sub>O mole fraction. However, five out of the six gas generation tests produced gas in which N<sub>2</sub>O exceeded N<sub>2</sub>.

The RGS-sampled tanks seemed to fall into two categories, one in which N<sub>2</sub>O was greater than 20 mol% (AN-104, U-103, SX-106, S-102, U-109, and SY-101), and the other a relatively low N<sub>2</sub>O group. Of the drill string-sampled tanks, only BY-110 contained more than 20 mol% N<sub>2</sub>O, while some tanks that showed high N<sub>2</sub>O in RGS samples had low N<sub>2</sub>O in drill string samples. Waste from Tanks A-101, S-102, S-106, and U-103 generated gas with more than 20 mol% N<sub>2</sub>O in laboratory tests.

Methane and non-methane hydrocarbons (NMHCs) made up the remaining low-solubility gas in the waste. RGS data indicate that CH<sub>4</sub> made up less than 1 mol% of the gas in 46 of the 75 samples taken. When non-methane hydrocarbons (C<sub>2</sub>H<sub>x</sub>, C<sub>3</sub>H<sub>x</sub>, and others) are included, the sum of CH<sub>4</sub>, C<sub>2</sub>H<sub>x</sub>, C<sub>3</sub>H<sub>x</sub>, and other hydrocarbons equaled or exceeded 3 mol% of the retained gas in 12 of the 75 RGS samples. These included five from AW-101 (with a maximum of 6 mol%), one from A-101 (5 mol%), one from AN-105 (3 mol%), one from AN-104 (3 mol%), one from BY-109 (3 mol%), two from SX-106 (the higher was 4 mol%), and one from AX-101 (3 mol%). Only one drill string sample, from BY-110, contained more than 1 mol% CH<sub>4</sub>. This same sample also contained a relatively high concentration, 2.9 mol%, of "other" gases. Of the six tanks whose waste was used in gas generation tests, only U-103 produced gas containing more than 3 mol% CH<sub>4</sub>.

The composition data obtained from headspace samples, RGS samples, drill string samples, and laboratory gas generation tests do not compare well. The reasons for the differences are not known. The  $H_2/N_2O$  ratio is used for comparison because it is the only composition property measured by all four methods. Part of the reason for differences may be that RGS methods showed in situ compositions, while headspace sample compositions often contain mixtures of gas from different layers (as may drill string samples). RGS samples show that gas composition can vary substantially from layer to layer.

Table 2.9 presents comparisons of the  $H_2/N_2O$  ratios in gas from RGS, drill string samples, SHMS, vapor grab samples, and laboratory gas generation tests on actual Hanford waste. Table 2.10 presents comparisons of the  $H_2/N_2$  ratios in gas from RGS, drill string samples, and gas generation tests. Certain specific features are evident from the data compiled in the tables.

**Table 2.9.** Comparison of  $H_2/N_2O$  Ratios from RGS, Drill String Samples, SHMS, Vapor Grab Samples and Laboratory Gas Generation Tests

Tank	RGS $H_2/N_2O$ (Tables 2.3, 2.5)	Drill String $H_2/N_2O$ (Table 2.7)	Average SHMS $H_2/N_2O$ (Table 2.1)	Average of other Headspace Samples $H_2/N_2O$ (Table 2.2)	Gas Generation Test $H_2/N_2O$ (Table 2.8)
A-101	Convective: $0.9 \pm 0.4$ – $2.5 \pm 1.2$ Nonconvective: $12 \pm 1.8$ – $13 \pm 1.7$		4.8	3.5	1.5
AN-103	Crust: $9.0 \pm 1.3$ – $10 \pm 1.4$ Convective: $2.3 \pm 2.1$ – $7.3 \pm 6.6$ Nonconvective: $15 \pm 2.8$ – $17 \pm 3.2$	9.2	11		
AN-104	Convective: $1.6 \pm 1.2$ – $5.0 \pm 4.1$ Nonconvective: $2.0 \pm 0.4$ – $2.4 \pm 0.5$	3.0	6.2		
AN-105	Convective: $1.9 \pm 1.3$ – $5.4 \pm 3.9$ Nonconvective: $3.9 \pm 0.5$ – $5.2 \pm 0.7$		5.3		
AW-101	Convective: $11 \pm 5.7$ – $31 \pm 16$ Nonconvective: $4.3 \pm 0.6$ – $6.5 \pm 1.0$	12.5	43		7.2
AX-101	Nonconvective: $5.5 \pm 0.7$ – $5.8 \pm 0.8$		9.6	>8.2	
BY-109	Nonconvective: $2.8 \pm 0.5$ – $3.0 \pm 0.5$	3.4	10		
S-102	Nonconvective: $1.0 \pm 0.1$ – $1.1 \pm 0.1$	1.7, 12.3	1.3	1.1	2.5
S-106	Nonconvective: $5.7 \pm 0.7$ – $7.1 \pm 1.0$	4.0, 6.3	>2		0.95
S-111	Convective: $2.9 \pm 3.3$ – $6.8 \pm 5.8$ Nonconvective: $6.0 \pm 1.3$ – $6.8 \pm 5.8$		7.3	8.1	
SX-106	Convective: $1.4 \pm 0.5$ – $6.4 \pm 2.4$ Nonconvective: $2.1 \pm 0.3$ – $2.2 \pm 0.3$	2.5	3.0	<7	
SY-101	Crust: $1.8 \pm 0.3$ Mixed slurry: $1.1 \pm 0.5$ – $1.9 \pm 0.8$		0.9		
U-103	Nonconvective: $0.6 \pm 0.05$		0.7	0.6	0.74
U-109	Nonconvective: $0.9 \pm 0.2$ – $1.0 \pm 0.2$	5.1, 1.2, 1.5	0.8	0.9	

**Table 2.10.** Comparison of H<sub>2</sub>/N<sub>2</sub> Ratios from RGS, Drill String Samples, SHMS, Vapor Grab Samples and Laboratory Gas Generation Tests

Tank	RGS H <sub>2</sub> /N <sub>2</sub> (Tables 2.3, 2.4)	Drill string H <sub>2</sub> /N <sub>2</sub> (Table 2.7)	Gas Generation Test H <sub>2</sub> /N <sub>2</sub> (Table 2.8)
A-101	Convective: 0.23±0.11–0.21±0.10 Nonconvective: 3.7±1.0–3.8±1.0		1.9
AN-103	Crust: 2.1±0.3–2.2±0.3 Convective: 0.3±0.3 Nonconvective: 1.8±0.3–1.9±0.3	1.0	
AN-104	Convective: 0.44±0.44 Nonconvective: 1.6±0.4–1.5±0.3	0.62	
AN-105	Convective: 0.44±0.38–0.39±0.34 Nonconvective: 2.5±0.5–2.6±0.5		
AW-101	Convective: 0.39±0.23–0.36±0.22 Nonconvective: 0.58±0.09	0.35	12
AX-101	Nonconvective: 3.8±0.7–3.6±0.6		
BY-109	Nonconvective: 1.8±0.4	0.57	
S-102	Nonconvective: 1.0±0.2	0.44, 0.61 <sup>(a)</sup>	5.0
S-106	Nonconvective: 2.5±0.4	0.87, 2.8	2.4
S-111	Convective: 0.07±0.08 Nonconvective: 3.1±1.0		
SX-106	Convective: 0.34±0.14–0.24±0.10 Nonconvective: 2.6±0.6–2.5±0.5	0.02	
SY-101	Crust: 1.3±0.3 Mixed slurry: 0.65±0.30		
U-103	Nonconvective: 0.6±0.06		0.80
U-109	Nonconvective: 0.54±0.11	0.81, 0.42, 0.38 <sup>(a)</sup>	
(a) Nitrogen purged.			

- In some cases (AW-101, A-101, and AN-103) the H<sub>2</sub>/N<sub>2</sub>O ratios measured by RGS are substantially different for the convective and nonconvective layers. The direction of the difference is not consistent.
- In almost all cases (S-111 and SY-101 are exceptions), the RGS-measured H<sub>2</sub>/N<sub>2</sub> ratio in the gas retained in the convective layer is between 0.2 and 0.45.
- The H<sub>2</sub>/N<sub>2</sub>O and H<sub>2</sub>/N<sub>2</sub> ratios of laboratory-generated gases do not match consistently those in gases measured in the waste or in the headspace. The direction of the difference is not consistent.
- The overall composition (H<sub>2</sub>/N<sub>2</sub>O/N<sub>2</sub>) of retained gas cannot be determined reliably from drill string samples because the H<sub>2</sub>/N<sub>2</sub> ratio cannot be measured meaningfully in high-air samples and the H<sub>2</sub>/N<sub>2</sub>O ratio does not correlate with values obtained by other methods.

RGS composition data, where available, are preferred to represent in situ gas composition because the measurements were made as nearly in situ as possible and contain all the constituents that affect flammability (Barker et al. 1999). However, ammonia released by evaporation must be added to RGS compositions of retained gas to represent the total release composition. Drill string and headspace data lack  $N_2$  measurements, which, in most retained gas, made up 20 to 50 mol% of the gas. It follows that  $N_2O$  and  $H_2$  together usually made up 50 to 80 mol% of the gas, so the absence of  $N_2$  data introduces about 25% uncertainty into estimates of the  $H_2$  flammability contribution in the headspace after a large release. A larger uncertainty comes from the fact that the  $H_2/N_2O$  ratios from headspace and drill string measurements often were within about 50% of the RGS values but could as easily differ by more than a factor of 2. Drill string  $H_2/N_2O$  ratios were usually closer to RGS values than the headspace measurements.

### 3.0 Flammability Limits

The principal flammable products of Hanford waste gas generation reactions are  $H_2$ ,  $N_2O$ , and  $NH_3$ .  $CH_4$  and other hydrocarbons are present to a much lesser extent. These substances have been observed in many laboratory studies with simulants and in laboratory studies with actual waste samples (Ashby et al. 1992, 1993, 1994; Barefield et al. 1995, 1996; Bryan and Pederson 1993, 1994, 1995; Camaioni et al. 1997; Delegard 1980; Johnson 1997; King et al. 1997; Meisel et al. 1991a, 1991b, 1992, 1993; Person 1996; Pederson and Bryan 1996). These same products are found in the headspace of actual waste tanks and in gas samples obtained using the RGS (Johnson 1997; Mahoney et al. 1999). This section summarizes experimental work investigating the flammability limits of these gases in the presence of air and  $N_2O$ , and applies the experimental results to assessing the flammability of the gases retained in selected tanks where gas compositions have been measured.

The flammability limits of a combustible mixture are those limiting compositions that will just support flame propagation when stimulated by an external ignition source. Although there is no widely accepted theoretical method for predicting flammability limits, there are a number of empirical rules and simple models, the results of which are summarized in Lewis and Von Elbe (1961) and updated by Hertzberg (1976). The fuel type, mixture properties, and mass diffusion of the deficient reactant are all factors in defining the limiting composition (Abbud-Madrid and Ronney 1990).

The primary reason for the existence of the flammability limit is heat loss (thermal radiation and conduction) from the hot products to the cooler surroundings (Spalding 1957). When the rate of heat loss exceeds the rate of heat generated by the chemical reactions, a flame cannot be supported. The composition at which a balance exists between heat loss and generation determines the flammability limit. Thermal radiation from infrared-active species  $CO_2$  and  $H_2O$  is one mechanism of heat loss from propagating flames. Thermal conduction to cold container walls is also a major heat loss mechanism in standard flammability tests (Coward and Jones 1952). Flammability limit measurements are therefore dependent on vessel size and geometry.

Flame propagation in fuel-rich mixtures is limited by oxidant availability and in lean mixtures by fuel availability. Thus the conditions of flammability are characterized by the upper (rich) and lower (lean) flammability limits of fuel gas concentration. In considering the flammability hazard associated with tank wastes, we are concerned more with the LFL than with the upper flammability limit (UFL). The LFL is determined by the gas composition, energy and location of the ignition source, direction of flame propagation relative to gravity, and whether the gas is quiescent or being mixed. The gas composition effects include those associated with different oxidants ( $O_2$  in air versus  $N_2O$ ) and those associated with inert constituents.

In a quiescent atmosphere, the transition to flammability is abrupt and the LFL is higher (i.e., more fuel is required) for downward flame propagation than for upward propagation because of the buoyancy of the heated gases. Thus, the upward propagation limit (the minimum fuel concentration required for a flame to propagate upward) is lower than the downward propagation limit (the minimum fuel concentration required for a flame to propagate downward). In a turbulent, well-mixed atmosphere the transition to flammability is more gradual, with combustion pressures increasing linearly as the fuel concentration increases from the upward

propagation limit. However, the flame propagation speed remains relatively low for concentrations above the upward propagation limit but below the downward propagation limit (Cashdollar et al. 1992). A very energetic ignition source has an effect similar to turbulence.

The flammability limits in air for both upward and downward propagation for H<sub>2</sub>, CH<sub>4</sub>, and NH<sub>3</sub> are summarized in Table 3.1.<sup>(a)</sup> The minimum amount of N<sub>2</sub> needed to inert a fuel/air mixture is also given. The corresponding limits for N<sub>2</sub>O are given in Table 3.2. The tables were compiled from experiments in which high ignition energies between 5 and 10 J were used.

**Table 3.1.** Flammability Limits of Hydrogen, Ammonia, and Methane in Air

Fuel	Lower Flammability Limit in Air (mol%)		Upper Flammability Limit in Air (mol%)	Minimum Nitrogen to Inert (mol%)
	Upward Propagation	Downward Propagation		
Hydrogen	4	8	75	70
Methane	5	5	15	37
Ammonia	15	18	28	17

**Table 3.2.** Flammability Limits of Hydrogen, Ammonia, and Methane in Nitrous Oxide

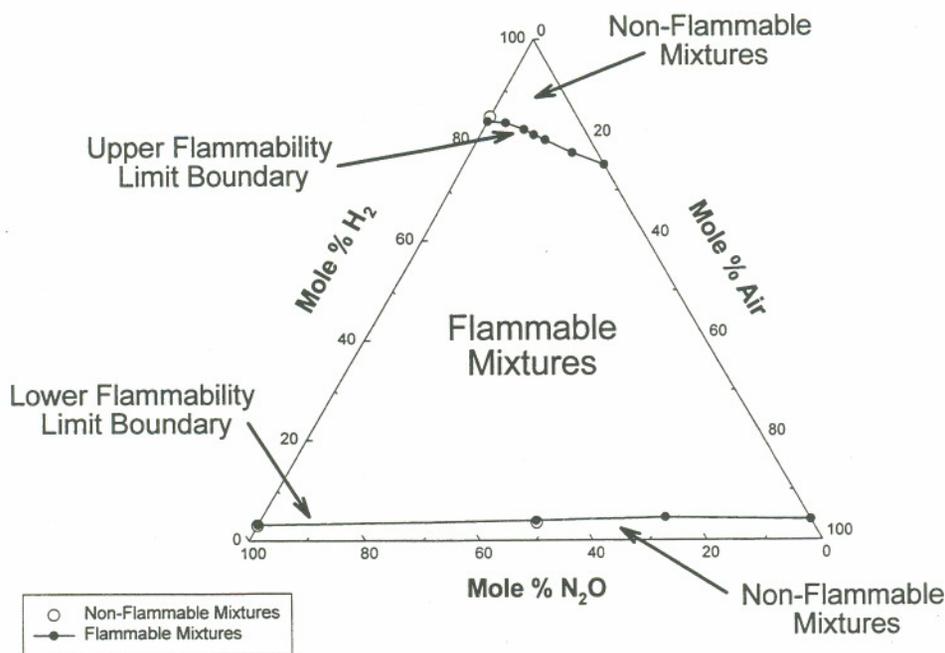
Fuel	Lower Flammability Limit in N <sub>2</sub> O (mol%)		Upper Flammability Limit in N <sub>2</sub> O (mol%)	Minimum Nitrogen to Inert (mol%)
	Upward Propagation	Downward Propagation		
Hydrogen	3	6	84	~60
Methane	2.8	5	50	70
Ammonia	2.6	6.8	71	60

### 3.1 Flammability of Hydrogen in Air and Nitrous Oxide

Hydrogen is the most abundant flammable gas in Hanford tanks. Methane and NH<sub>3</sub> also contribute to mixture flammability, though to a much smaller extent in the typical waste tank scenario. Other organic gases may also be present at very small concentrations but have no practical effect on flammability (Hedengren et al. 1997).

The limits for flammability for H<sub>2</sub>-N<sub>2</sub>O-air mixtures are shown in Figure 3.1 (Coward and Jones 1952; Smith and Linnett 1953; Zabetakis 1965). *Note that the hydrogen LFLs given in Tables 3.1 and 3.2 represent the end points of Figure 3.1.*

(a) Pfahl U and JE Shepherd. April 29, 1997. *Flammability and Flame Propagation in H<sub>2</sub>-N<sub>2</sub>O-CH<sub>4</sub>-NH<sub>3</sub>-O<sub>2</sub>-N<sub>2</sub> Mixtures*. Report FM97-4, Explosion Dynamics Laboratory, Pasadena, California.



**Figure 3.1.** Limits of Flammability of the System  $\text{H}_2\text{-N}_2\text{O-Air}$  at  $28^\circ\text{C}$  and 1 atm (Coward and Jones 1952; Smith and Linnett 1953; Zabetakis 1965)

The LFLs of  $\text{H}_2$  in air- $\text{N}_2\text{O}$  mixtures were recently investigated in detail by Breshears et al. (1997), Cashdollar et al. (1992), and others.<sup>(a)</sup> Their findings were in agreement with the earlier work. These studies all confirm that the LFL for  $\text{H}_2$  (8 to 9 mol%  $\text{H}_2$ , quiescent mixtures) does not change significantly when air is replaced by  $\text{N}_2\text{O}$  as the oxidant (Figure 3.1), even though the reaction of  $\text{H}_2$  with  $\text{N}_2\text{O}$  (Eq. 3.1) is significantly more exothermic than with  $\text{O}_2$  (Eq. 3.2).



Analysis by Breshears et al. (1997) shows these results to be consistent with a chemical kinetic model in which (at atmospheric pressures and temperatures below about 1000K) the dominant channel for reaction of hydrogen atoms is formation of  $\text{HO}_2$ , the hydroperoxy radical:



(a) Ross MC and JE Shepherd. July 9, 1996. *Lean Combustion Characteristics of Hydrogen-Nitrous Oxide-Ammonia Mixtures in Air. Part I.* Report FM96-4, Explosion Dynamics Laboratory, Pasadena, California.

Several subsequent reactions may occur, including the sequence



Summation of Eq. (3.3), (3.4) and twice (3.5) yields twice (3.1). Reactions (3.3) through (3.5) compose a *straight chain* mechanism in which the net atom and radical concentration is conserved. However, addition reaction channels such as



result in a net loss of atoms and radicals such that the flame propagation solely by a straight chain mechanism is considered to be highly improbable.

At temperatures above about 1000K, the *chain branching* reaction



becomes the dominant hydrogen atom reaction path, followed by reaction (3.5) and a second *chain branching* reaction:



Summing Eq. (3.7), (3.8), and (3.5) yields



Now, two additional atoms and radicals are produced for each  $\text{O}_2$  molecule consumed, and the net chain branching reaction can overcome the normal loss mechanisms.

The LFL of 8 to 9 mol%  $\text{H}_2$  (quiescent) for  $\text{H}_2$ -air mixtures corresponds very closely to the amount of  $\text{H}_2$  combustion required to reach a temperature of about 1000K, i.e., to a temperature above which the chain branching reaction dominates. Further, the dominant channel for reaction of  $\text{N}_2\text{O}$  in this system is



which, summed with reaction (3.5), yields (3.2).

Rate constants for reactions (3.7) and (3.10) are very nearly equal at temperatures above about 700K, with the result that  $\text{N}_2\text{O}$  begins to react with  $\text{H}_2$  only for initial  $\text{H}_2$  mole fractions that are already above the LFL for  $\text{H}_2$ -air mixtures. Moreover, a very energetic ignition source is

required to bring the temperature to the point where the reaction can occur. Below that temperature, N<sub>2</sub>O acts as a diluent. An H<sub>2</sub> mole fraction of 0.10 or more is required to fully react the N<sub>2</sub>O.<sup>(a)</sup>

No information was found concerning the LFL of H<sub>2</sub> in N<sub>2</sub>O as a function of the ignition energy. It is known that, at very high ignition energies (i.e., on the order of 5000 J), N<sub>2</sub>O decomposes even in the absence of fuel.

### 3.2 Flammability of Ammonia in Air and Nitrous Oxide

Ammonia is a pervasive constituent of wastes stored at Hanford (Mahoney et al. 1999). Ammonia is difficult to ignite and was not even considered flammable as recently as 1951 (Buckley and Husa 1962). Ammonia has a very low flame speed and requires relatively high concentrations to support combustion (its LFL in air is about 15 mol%) and relatively high-energy ignition sources to initiate combustion (8 to 680 mJ compared with about 0.01 mJ for CH<sub>4</sub>).<sup>(a)</sup> These minimum ignition energies pertain to fuel-rich conditions; in general, higher ignition energies are required near the LFL.

Because NH<sub>3</sub> is used extensively as a refrigerant, the combustion of NH<sub>3</sub>-air mixtures has been well studied (Fenton et al. 1995). Accepted limits for NH<sub>3</sub>-air flammabilities are scattered around 15 mol% for upward and 18 mol% for downward propagation (Ronney 1987, Table 3.2). Unlike H<sub>2</sub>, the UFL of NH<sub>3</sub> is reduced when N<sub>2</sub>O is substituted for air as the oxidant. The presence of 8 mol% N<sub>2</sub>O (in air) decreased the LFL of NH<sub>3</sub> from 18 mol% to 15 mol%.<sup>(b)</sup> While the mechanism for the reaction of NH<sub>3</sub> with N<sub>2</sub>O was not captured by a simple kinetic model (Breshears et al. 1997), factors contributing to the lowering of the LFL of NH<sub>3</sub> by addition of N<sub>2</sub>O were investigated and discussed.<sup>(b)</sup>

Apparently, the differences in the effect of N<sub>2</sub>O on H<sub>2</sub> and NH<sub>3</sub> mixtures are due to several factors. Temperature is the main factor.<sup>(a)</sup> Lean H<sub>2</sub>-air flames have a very low product temperature (<1100K) and N<sub>2</sub>O decomposes very slowly under these conditions. In contrast, the product temperature of lean NH<sub>3</sub>-air flames is much higher, and at the higher temperature N<sub>2</sub>O decomposes readily. In mixtures where the only oxidant is N<sub>2</sub>O (no air), the decomposition of N<sub>2</sub>O is stoichiometrically complete. In addition, there appear to be reactive intermediates characteristic of nitrogen chemistry that play a role in NH<sub>3</sub>-N<sub>2</sub>O combustion and are not factors for H<sub>2</sub>-N<sub>2</sub>O combustion.

The differences in radicals and reactive intermediates are one explanation offered for the observation that the NH<sub>3</sub>-N<sub>2</sub>O LFL is lower than the NH<sub>3</sub>-air LFL. At high temperatures (>1300K), the primary reaction responsible for the decomposition of NH<sub>3</sub> is

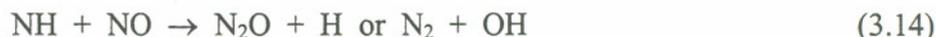
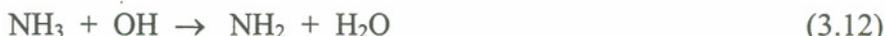


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(a) Pfahl U and JE Shepherd. April 29, 1997. *Flammability and Flame Propagation in H<sub>2</sub>-N<sub>2</sub>O-CH<sub>4</sub>-NH<sub>3</sub>-O<sub>2</sub>-N<sub>2</sub> Mixtures*. Report FM97-4, Explosion Dynamics Laboratory, Pasadena, California.

(b) Ross MC and JE Shepherd, *Lean Combustion Characteristics of Hydrogen-Nitrous Oxide-Ammonia Mixtures in Air. Part I*. Report FM96-4, Explosion Dynamics Laboratory, Pasadena, California.

In mixtures of H<sub>2</sub> and NH<sub>3</sub>, this reaction can proceed simultaneously with reaction (3.5) without interference.<sup>(a)</sup> It is known that at lower temperatures (<1100K) the chain branching mechanism is not important, but reaction (3.3) is the primary path for hydrogen atom reaction. The primary mechanism for NH<sub>3</sub> consumption in lower temperature flames appears to be



The decrease in the LFL of NH<sub>3</sub> from 15 mol% in air to an estimated 2.6 mol% in N<sub>2</sub>O may be due to the decomposition of N<sub>2</sub>O in the high-temperature NH<sub>3</sub> flames following the reactions



and



The flammability limits of NH<sub>3</sub> in N<sub>2</sub>O are strong functions of the ignition energy.<sup>(b)</sup> At an ignition energy of 0.04 J, the LFL is 11.5 mol% NH<sub>3</sub> rather than the 5.2 mol% found for ignition energies of 1 J or more. The UFL decreases from 67.5 mol% at 8 J ignition energy to 54 mol% at 0.04 J ignition energy. Both the LFL and the UFL are independent of the ignition energy for ignition energies between 1 J and 10 J and may remain independent up to 100 J or more. For a mixture of 40% NH<sub>3</sub>-60% N<sub>2</sub>O (well above the LFL), a minimum ignition energy of 0.07 mJ was measured (Calcote et al. 1952); it is typical for minimum ignition energies to be found in fuel-rich mixtures.

### 3.3 Effect of Water Vapor on Ammonia-Air LFL

Water vapor raises the LFL of NH<sub>3</sub>-air mixtures. If the water vapor mole fraction exceeds 8 to 11 mol%, the mixture is not flammable. For comparison, 17 to 19 mol% N<sub>2</sub> is required to inert a mixture whose other components are NH<sub>3</sub> and air (at any NH<sub>3</sub>/air ratio). The amount of water vapor necessary to prevent combustion increases from about 8 mol% at 25°C to about 9 mol% at 44°C, and to about 11 mol% at 80°C (Fenton et al. 1995). This is important because it is possible for the equilibrium headspace humidity to exceed this value at moderate temperatures.

The equilibrium water vapor fraction as a function of temperature and the flammability range for NH<sub>3</sub>-air-water vapor mixtures are shown on Figure 3.2. The equilibrium water vapor fraction above pure water (CRC 1990) exceeds the inerting value at 41° to 48°C (106° to 118°F). The

(a) Ross MC and JE Shepherd. July 9, 1996. *Lean Combustion Characteristics of Hydrogen-Nitrous Oxide-Ammonia Mixtures in Air. Part I.* Report FM96-4, Explosion Dynamics Laboratory, Pasadena, California.

(b) Pfahl U and JE Shepherd. April 29, 1997. *Flammability and Flame Propagation in H<sub>2</sub>-N<sub>2</sub>O-CH<sub>4</sub>-NH<sub>3</sub>-O<sub>2</sub>-N<sub>2</sub> Mixtures.* Report FM97-4, Explosion Dynamics Laboratory, Pasadena, California.

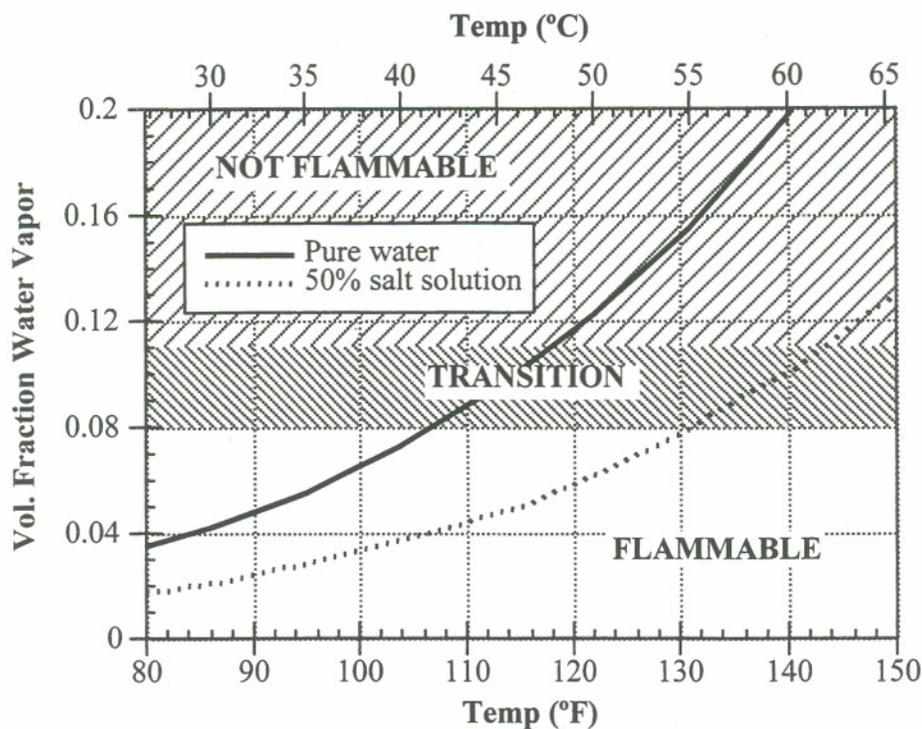


Figure 3.2. Equilibrium Water Vapor Fraction and Ammonia Flammability

vapor pressure over a salt solution containing 50 mol% water (shown by the dashed line) is about half the vapor pressure over pure water based on a correlation developed by Mahoney and Trent (1995) from SY-101 chemical simulant data. This increases the temperature at which the mixture is inert to about 55° to 61°C. However, even if the temperature were sufficiently high, water vapor cannot confidently be claimed to prevent combustion because the mixture will contain H<sub>2</sub>, which is not effectively inerted by the relatively low mole fractions of water vapor present in tank headspaces.<sup>(a)</sup>

### 3.4 Methane Flammability in Air and Nitrous Oxide

The flammability of CH<sub>4</sub> and other hydrocarbons in air has been studied extensively. The LFL for CH<sub>4</sub> in air is 5.0 mol% (Zabetakis 1965). The LFL for CH<sub>4</sub> decreases to 4.8 mol% in a mixture containing 10 mol% N<sub>2</sub>O in air. The flammability limits of ternary mixtures of CH<sub>4</sub>-N<sub>2</sub>O-N<sub>2</sub> have been investigated. The LFL for CH<sub>4</sub> in N<sub>2</sub>O is 2.8 mol%,<sup>(b)</sup> and the UFL is 40 to 50 mol%. Adding 70 mol% N<sub>2</sub> will inert the mixture.

(a) Ross MC and JE Shepherd. July 9, 1996. *Lean Combustion Characteristics of Hydrogen-Nitrous Oxide-Ammonia Mixtures in Air. Part I.* Report FM96-4, Explosion Dynamics Laboratory, Pasadena, California.

(b) Pfahl U and JE Shepherd. April 29, 1997. *Flammability and Flame Propagation in H<sub>2</sub>-N<sub>2</sub>O-CH<sub>4</sub>-NH<sub>3</sub>-O<sub>2</sub>-N<sub>2</sub> Mixtures.* Report FM97-4, Explosion Dynamics Laboratory, Pasadena, California.

The characteristics of CH<sub>4</sub>-air-N<sub>2</sub>O mixtures differ from the results of H<sub>2</sub>-air-N<sub>2</sub>O mixtures described above. Methane-air-N<sub>2</sub>O mixtures behave similarly to NH<sub>3</sub>-air-N<sub>2</sub>O mixtures in that the LFL for CH<sub>4</sub> is decreased when N<sub>2</sub>O replaces air as the oxidant within the ternary system. The heats of combustion of CH<sub>4</sub>-air-N<sub>2</sub>O mixtures are higher than those of the H<sub>2</sub>-air-N<sub>2</sub>O system. Therefore, the flame temperature at the flammability limit will be much higher for CH<sub>4</sub> than for H<sub>2</sub>. It follows that the rate of thermal decomposition of N<sub>2</sub>O following reaction (3-15) with an Arrhenius activation energy of approximately 60 kcal/mol (Breshears 1995) will be high at the CH<sub>4</sub> LFL, and decomposition will always occur.<sup>(a)</sup> This was also observed for the NH<sub>3</sub>-air-N<sub>2</sub>O system with significant N<sub>2</sub>O decomposition occurring near the lean limit of mixtures containing large fractions of NH<sub>3</sub>.<sup>(a)</sup>

Although CH<sub>4</sub> and N<sub>2</sub>O burn readily, the amount of CH<sub>4</sub> in tank wastes is small. Other hydrocarbons are not a flammability concern, even though their LFLs are even lower than CH<sub>4</sub>, because their concentrations are typically much lower even than those of CH<sub>4</sub>.

The LFL of CH<sub>4</sub> in N<sub>2</sub>O is a strong function of the ignition energy.<sup>(a)</sup> At an ignition energy of 0.04 J, the LFL is 4.8 mol% CH<sub>4</sub>, but it decreases to 2.8 mol% for ignition energies of 1 J or more. The LFL is independent of the ignition energy for ignition energies between 1 J and 10 J and may remain independent up to 100 J or more.

### 3.5 Flammability of Hydrogen, Methane, and Ammonia Mixtures in Air

Experiments (Cashdollar et al. 1992) have shown that the LFL of a mixture of H<sub>2</sub>, NH<sub>3</sub>, and CH<sub>4</sub> in an air-N<sub>2</sub>O atmosphere can be computed from the LFL of each fuel using LeChatelier's linear mixing law:

$$\frac{1}{[\text{MIX}]_{\text{LFL}}} = \frac{[\text{H}_2]_{\text{MIX}}}{[\text{H}_2]_{\text{LFL}}} + \frac{[\text{CH}_4]_{\text{MIX}}}{[\text{CH}_4]_{\text{LFL}}} + \frac{[\text{NH}_3]_{\text{MIX}}}{[\text{NH}_3]_{\text{LFL}}} \quad (3.17)$$

where [MIX]<sub>LFL</sub> = concentration of the gas mixture at the LFL in air  
 [gas]<sub>MIX</sub> = concentration of the gas composing the fuel mixture  
 [gas]<sub>LFL</sub> = concentration of the pure gas at the LFL in air.

The concentration of H<sub>2</sub> is usually monitored to detect flammable conditions. The H<sub>2</sub> concentration at which a mixture of H<sub>2</sub> with other gases is flammable is given by

$$[\text{H}_2]_{\text{MIX,LFL}} = X_{\text{H}_2}[\text{MIX}]_{\text{LFL}} \quad (3.18)$$

where  $X_{\text{H}_2}$  = mole fraction of H<sub>2</sub>. In a waste tank 23 m (75 ft) in diameter, the headspace would generally be expected to be turbulent, and the upward propagation flammability limit is appropriate for determining whether ignition is possible.

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(a) Ross MC and JE Shepherd. July 9, 1996. *Lean Combustion Characteristics of Hydrogen-Nitrous Oxide-Ammonia Mixtures in Air. Part I.* Report FM96-4, Explosion Dynamics Laboratory, Pasadena, California.

Table 3.3 lists the waste gas fractions required to reach the LFL for gas from the gas-retaining layer(s) of the RGS-sampled tanks, calculated from Eq. (3.17) and based on the RGS compositions given in Section 2 for the low-gas solubility bound. The waste gas fraction is the mole fraction of waste gas in the mixture of air and waste gas. The fractions in Table 3.3 do not include the extra ammonia that could be released by evaporation and are not conservative for significant evaporative releases. Because of its higher LFL, ammonia has only 27% as much effect on flammability as hydrogen—it takes 3.75 mol% NH<sub>3</sub> to have as much effect as 1 mol% H<sub>2</sub>. The lower effectiveness of ammonia makes up, to some extent, for the greater uncertainty in its concentration (compared with hydrogen).

**Table 3.3.** Waste Gas Fractions at the LFL in Air/Gas Mixtures

Source of Gas	Composition (mol%)	Fraction <sup>(a)</sup> of Gas in Air/Gas Mixture to Reach LFL (mol%)	Fraction <sup>(a)</sup> of H <sub>2</sub> in Mixture at LFL (mol%)
A-101 nonconvective	19% N <sub>2</sub> , 70% H <sub>2</sub> , 4.8% NH <sub>3</sub> , 5.7% N <sub>2</sub> O, 0.7% CH <sub>4</sub>	5.6	3.9
AN-103 crust	29% N <sub>2</sub> , 62% H <sub>2</sub> , 1.8% NH <sub>3</sub> , 6.9% N <sub>2</sub> O, 0.6% CH <sub>4</sub>	6.3	3.9
AN-103 nonconvective	33% N <sub>2</sub> , 61% H <sub>2</sub> , 0.9% NH <sub>3</sub> , 4.2% N <sub>2</sub> O, 0.6% CH <sub>4</sub>	6.5	4.0
AN-104 nonconvective	29% N <sub>2</sub> , 45% H <sub>2</sub> , 1.4% NH <sub>3</sub> , 23% N <sub>2</sub> O, 0.9% CH <sub>4</sub>	8.6	3.9
AN-105 nonconvective	24% N <sub>2</sub> , 59% H <sub>2</sub> , 0.6% NH <sub>3</sub> , 15% N <sub>2</sub> O, 0.7% CH <sub>4</sub>	6.7	3.9
AW-101 nonconvective	55% N <sub>2</sub> , 32% H <sub>2</sub> , 0.9% NH <sub>3</sub> , 7.5% N <sub>2</sub> O, 1.7% CH <sub>4</sub>	12	3.8
AX-101 nonconvective	16% N <sub>2</sub> , 60% H <sub>2</sub> , 9.1% NH <sub>3</sub> , 11% N <sub>2</sub> O, 2.4% CH <sub>4</sub>	6.2	3.7
BY-109 nonconvective	28% N <sub>2</sub> , 50% H <sub>2</sub> , 0.3% NH <sub>3</sub> , 18% N <sub>2</sub> O, 0.8% CH <sub>4</sub>	7.8	3.9
S-102 nonconvective	32% N <sub>2</sub> , 33% H <sub>2</sub> , 1.0% NH <sub>3</sub> , 33% N <sub>2</sub> O, 0.4% CH <sub>4</sub>	12	4.0
S-106 nonconvective	25% N <sub>2</sub> , 63% H <sub>2</sub> , 0.4% NH <sub>3</sub> , 11% N <sub>2</sub> O, 0.3% CH <sub>4</sub>	6.3	4.0
S-111 nonconvective	21% N <sub>2</sub> , 66% H <sub>2</sub> , 1.0% NH <sub>3</sub> , 11% N <sub>2</sub> O, 0.5% CH <sub>4</sub>	6.0	4.0
SX-106 nonconvective	20% N <sub>2</sub> , 51% H <sub>2</sub> , 4.4% NH <sub>3</sub> , 24% N <sub>2</sub> O, 0.6% CH <sub>4</sub>	7.5	3.8
SY-101 crust	27% N <sub>2</sub> , 34% H <sub>2</sub> , 19% NH <sub>3</sub> , 19% N <sub>2</sub> O, 0.6% CH <sub>4</sub>	10	3.4
SY-101 mixed slurry	40% N <sub>2</sub> , 26% H <sub>2</sub> , 7.1% NH <sub>3</sub> , 24% N <sub>2</sub> O, 1.3% CH <sub>4</sub>	14	3.6
U-103 nonconvective	36% N <sub>2</sub> , 23% H <sub>2</sub> , 0.6% NH <sub>3</sub> , 40% N <sub>2</sub> O, 0.4% CH <sub>4</sub>	16	3.7
U-109 nonconvective	46% N <sub>2</sub> , 25% H <sub>2</sub> , 0.9% NH <sub>3</sub> , 27% N <sub>2</sub> O, 0.7% CH <sub>4</sub>	15	3.8

(a) The gas fractions are based on retained gas composition and are higher (less conservative) than would be required to reach LFL if a substantial amount of ammonia were released by evaporation as well as by bubbles.

The effects of inerting by the  $N_2$  in the waste gas were not considered, giving somewhat conservative (underestimated) waste-gas fractions. However, little inerting effect is expected at the low fractions of waste gas that are required to reach the LFL.

### 3.6 Self-Flammability of Hydrogen, Methane, Ammonia, and Nitrous Oxide Mixtures

Given the right gas composition and an adequate ignition source, combustion could in theory occur in gas in the pores of a matrix. Thus the self-flammability of the waste gas alone, in the absence of air, must be considered. Under airless conditions,  $N_2O$  serves as the sole oxidant, and  $N_2$  may in some cases inert the retained gas.

As the fraction of inert gas (in this case,  $N_2$ ) increases in a mixture, the LFL typically increases slightly and the UFL decreases steeply. At a fraction of the inert gas that is known as the inerting concentration, the UFL decreases to the point that it is equal to the LFL. When the inert gas increases above the inerting concentration, combustion cannot occur.

The LFL of  $NH_3$  in  $N_2O$  increases almost linearly from 5.2 mol% (with no  $N_2$  present) to 11.5 mol% when the inerting concentration of 60 mol%  $N_2$  is present.<sup>(a)</sup> The UFL of  $NH_3$  in  $N_2O$  at 60 mol%  $N_2$  is 16 mol%  $NH_3$ , somewhat higher than the LFL. When no  $N_2$  is present, the UFL in  $N_2O$  of  $NH_3$  is 71 mol%.

The LFL of  $CH_4$  in  $N_2O$  increases smoothly, though not quite linearly, from 2.8 mol% (with no  $N_2$  present) to 6.3 mol% when 70 mol%  $N_2$  is present (inerting concentration). The UFL of  $CH_4$  in  $N_2O$  is 50 mol% in the absence of  $N_2$  and 6.3 mol%  $CH_4$  (at the inerting concentration).

The effect of  $N_2$  on the UFL and LFL of  $H_2$  in  $N_2O$  is not known. Nitrogen has little effect on the LFL of  $H_2$  in air; the LFL in air is 4 mol% with no excess  $N_2$  and at the inerting concentration of  $N_2$ .<sup>(b)</sup> If the same behavior is assumed for  $H_2$  in  $N_2O$ , the LFL of  $H_2$  in  $N_2O$  would be 3 mol% at the inerting concentration of  $N_2$  (about 60 mol%  $N_2$ ). Because the LFLs and UFLs of the fuel gases in  $N_2O$  vary nearly linearly with  $N_2$  concentration, they can be expressed as

$$[H_2]_{LFL} = 0.03 \quad (3.19)$$

$$[H_2]_{UFL} = 0.84 - (0.84 - 0.03)[N_2]/0.6 \quad (3.20)$$

$$[NH_3]_{LFL} = 0.052 + (0.115 - 0.052)[N_2]/0.6 \quad (3.21)$$

$$[NH_3]_{UFL} = 0.71 - (0.71 - 0.16)[N_2]/0.6 \quad (3.22)$$

$$[CH_4]_{LFL} = 0.028 + (0.063 - 0.028)[N_2]/0.7 \quad (3.23)$$

$$[CH_4]_{UFL} = 0.50 - (0.50 - 0.063)[N_2]/0.7 \quad (3.24)$$

(a) Pfahl U and JE Shepherd April 29, 1997. *Flammability and Flame Propagation in  $H_2$ - $N_2O$ - $CH_4$ - $NH_3$ - $O_2$ - $N_2$  Mixtures*. Report FM97-4, Explosion Dynamics Laboratory, Pasadena, California.

(b) Shepherd JE. April 30, 1997. "Combustion and Explosion Experiments at Caltech." California Institute of Technology presentation to SCOPE Panel, Richland, Washington.

The validity of Eq. (3.19) through (3.24) was tested by applying them to gas mixtures whose flammability was measured by other researchers.<sup>(a)</sup> Mixtures that contained, respectively, 40% H<sub>2</sub>/20% CH<sub>4</sub>/40% N<sub>2</sub>O and 35% H<sub>2</sub>/20% NH<sub>3</sub>/10% CH<sub>4</sub>/35% N<sub>2</sub>O were self-flammable, as predicted. A mixture that contained 29% H<sub>2</sub>/11% NH<sub>3</sub>/1% CH<sub>4</sub>/24% N<sub>2</sub>O/35% N<sub>2</sub> (intended to simulate SY-101 retained gas) was predicted to be self-flammable but was not. It became flammable when mixed with less than 10 mol% air.

Table 3.4 contains an assessment of the self-flammability of the gases from the nonconvective layer(s) of each of the RGS-sampled tanks based on Eq. (3.19) through (3.24) and the RGS retained gas compositions given in Section 2 for the low gas-solubility bound. In 11 of

**Table 3.4.** Waste Gas Fractions at the LFL in Air/Gas Mixtures

Source of Gas	Composition	Flammability
A-101 nonconvective	19% N <sub>2</sub> , 70% H <sub>2</sub> , 4.8% NH <sub>3</sub> , 5.7% N <sub>2</sub> O, 0.7% CH <sub>4</sub>	H <sub>2</sub> is 120% of the UFL; NH <sub>3</sub> and CH <sub>4</sub> together are 85% of the LFL; not self-flammable
AN-103 crust	29% N <sub>2</sub> , 62% H <sub>2</sub> , 1.8% NH <sub>3</sub> , 6.9% N <sub>2</sub> O, 0.6% CH <sub>4</sub>	H <sub>2</sub> is 140% of the UFL; NH <sub>3</sub> and CH <sub>4</sub> together are 36% of the LFL; not self-flammable
AN-103 nonconvective	33% N <sub>2</sub> , 61% H <sub>2</sub> , 0.9% NH <sub>3</sub> , 4.2% N <sub>2</sub> O, 0.6% CH <sub>4</sub>	H <sub>2</sub> is 160% of the UFL; NH <sub>3</sub> and CH <sub>4</sub> together are 24% of the LFL; not self-flammable
AN-104 nonconvective	29% N <sub>2</sub> , 45% H <sub>2</sub> , 1.4% NH <sub>3</sub> , 23% N <sub>2</sub> O, 0.9% CH <sub>4</sub>	H <sub>2</sub> is 100% of the UFL; NH <sub>3</sub> and CH <sub>4</sub> together are 38% of the LFL; possibly self-flammable
AN-105 nonconvective	24% N <sub>2</sub> , 59% H <sub>2</sub> , 0.6% NH <sub>3</sub> , 15% N <sub>2</sub> O, 0.7% CH <sub>4</sub>	H <sub>2</sub> is 110% of the UFL; NH <sub>3</sub> and CH <sub>4</sub> together are 25% of the LFL; not self-flammable
AW-101 nonconvective	55% N <sub>2</sub> , 32% H <sub>2</sub> , 0.9% NH <sub>3</sub> , 7.5% N <sub>2</sub> O, 1.7% CH <sub>4</sub>	H <sub>2</sub> is 330% of the UFL; NH <sub>3</sub> and CH <sub>4</sub> together are 39% of the LFL; not self-flammable
AX-101 nonconvective	16% N <sub>2</sub> , 60% H <sub>2</sub> , 9.1% NH <sub>3</sub> , 11% N <sub>2</sub> O, 2.4% CH <sub>4</sub>	H <sub>2</sub> is 100% of the UFL; NH <sub>3</sub> is 130% of the LFL; possibly self-flammable
BY-109 nonconvective	28% N <sub>2</sub> , 50% H <sub>2</sub> , 0.3% NH <sub>3</sub> , 18% N <sub>2</sub> O, 0.8% CH <sub>4</sub>	H <sub>2</sub> is 110% of the UFL; NH <sub>3</sub> and CH <sub>4</sub> together are 23% of the LFL; not self-flammable
S-102 nonconvective	32% N <sub>2</sub> , 33% H <sub>2</sub> , 1.0% NH <sub>3</sub> , 33% N <sub>2</sub> O, 0.4% CH <sub>4</sub>	H <sub>2</sub> is 80% of the UFL; NH <sub>3</sub> and CH <sub>4</sub> together are 21% of the LFL; self-flammable
S-106 nonconvective	25% N <sub>2</sub> , 63% H <sub>2</sub> , 0.4% NH <sub>3</sub> , 11% N <sub>2</sub> O, 0.3% CH <sub>4</sub>	H <sub>2</sub> is 120% of the UFL; NH <sub>3</sub> and CH <sub>4</sub> together are 12% of the LFL; not self-flammable
S-111 nonconvective	21% N <sub>2</sub> , 66% H <sub>2</sub> , 1.0% NH <sub>3</sub> , 11% N <sub>2</sub> O, 0.5% CH <sub>4</sub>	H <sub>2</sub> is 120% of the UFL; NH <sub>3</sub> and CH <sub>4</sub> together are 26% of the LFL; not self-flammable
SX-106 nonconvective	20% N <sub>2</sub> , 51% H <sub>2</sub> , 4.4% NH <sub>3</sub> , 24% N <sub>2</sub> O, 0.6% CH <sub>4</sub>	H <sub>2</sub> is 90% of the UFL; NH <sub>3</sub> and CH <sub>4</sub> together are 76% of the LFL; self-flammable
SY-101 crust	27% N <sub>2</sub> , 34% H <sub>2</sub> , 19% NH <sub>3</sub> , 19% N <sub>2</sub> O, 0.6% CH <sub>4</sub>	H <sub>2</sub> is 70% of the UFL; NH <sub>3</sub> is 240% of the LFL; self-flammable
U-103 nonconvective	36% N <sub>2</sub> , 23% H <sub>2</sub> , 0.6% NH <sub>3</sub> , 40% N <sub>2</sub> O, 0.4% CH <sub>4</sub>	H <sub>2</sub> is 65% of the UFL; NH <sub>3</sub> and CH <sub>4</sub> together are 15% of the LFL; self-flammable
U-109 nonconvective	46% N <sub>2</sub> , 25% H <sub>2</sub> , 0.9% NH <sub>3</sub> , 27% N <sub>2</sub> O, 0.7% CH <sub>4</sub>	H <sub>2</sub> is 110% of the UFL; NH <sub>3</sub> and CH <sub>4</sub> together are 23% of the LFL; not self-flammable

(a) Pfahl U and JE Shepherd. April 29, 1997. *Flammability and Flame Propagation in H<sub>2</sub>-N<sub>2</sub>O-CH<sub>4</sub>-NH<sub>3</sub>-O<sub>2</sub>-N<sub>2</sub> Mixtures*. Report FM97-4, Explosion Dynamics Laboratory, Pasadena, California.

the 15 cases, the H<sub>2</sub> mole fraction is equal to or greater than the UFL for the mixture. Because the mixture UFLs are not precisely defined, any mixture in which H<sub>2</sub> is less than 110% of the UFL is considered potentially self-flammable. Of the 15 wastes listed, nine retain gas that contains too little oxidant to be self-flammable.

The gases in the AN-104 and AX-101 nonconvective layers contain borderline concentrations of H<sub>2</sub> and may be self-flammable. However, the total fuel/oxidant ratio (including NH<sub>3</sub> and CH<sub>4</sub> as fuels) may be too large to allow combustion, particularly for the higher-NH<sub>3</sub> AX-101 waste. In the four remaining wastes (S-102, SX-106, SY-101, and U-103), H<sub>2</sub> is below the UFL and the gas is self-flammable. Considering that the UFL model in Eq. (3.19) through (3.24) overpredicts the UFL, thereby increasing the range of mixtures that are modeled as self-flammable, it is possible that some of the waste gases would not actually exhibit self-flammability even though it was predicted.

It cannot be overstressed that this self-flammability assessment considers only the intrinsic properties of the gas mixture, as if the gas were contained in a large vessel rather than in the pores, bubbles, or fissures in the waste matrix. The assessment does not consider the amount of ignition energy required to initiate combustion. Many of the data used to support the flammability estimate were obtained with ignition energies of 8 J or more, which would be unusually high energies for accidental sparks (20 mJ energies are typically seen in static sparks on clothing). The self-flammability assessment also does not consider whether interconnected pathways exist in the waste to permit flame propagation or whether heat loss to the walls of a small passage would prevent propagation. All of these limiting factors make flame propagation within the waste implausible, as is discussed in Section 4.

### 3.7 Conclusions

The following conclusions were drawn from this flammability research:

- The upward propagation LFL for a gas mixture obeyed LeChatelier's linear mixing law (Eq. 3.17 and 3.18)
- Nitrous oxide is inert in H<sub>2</sub> combustion below the downward propagation limit of 8 mol% H<sub>2</sub>
- Nitrous oxide participates in NH<sub>3</sub> and CH<sub>4</sub> combustion at their LFLs and affects both the UFL and the LFL
- Neither water vapor nor waste-gas N<sub>2</sub> can be counted on to prevent combustion in tank headspaces or in the waste itself in every tank
- In many wastes, there is not enough N<sub>2</sub>O to support combustion in the absence of air.

Overall, H<sub>2</sub> dominates the flammability issue even at relatively low concentrations, and flammability limits for H<sub>2</sub> in the presence of other fuel gases can be readily calculated.

## 4.0 Deflagration and Detonation

When a gas mixture is exposed to an ignition source that is energetic enough to initiate combustion, the resulting flame propagation may be either by subsonic deflagration or supersonic detonation. This section reviews experimental studies of the outcome of ignition of such gas mixtures and discusses the parameters that govern whether deflagration or detonation could result.

Deflagration is subsonic flame propagation away from the ignition site. The peak pressure caused by combustion is essentially uniform throughout an enclosure and depends on the completeness of combustion, the resulting adiabatic flame temperature, and the increase (or decrease) in the number of moles of gas dictated by the combustion reaction stoichiometry. A detonation involves supersonic flame propagation. The peak pressure of a detonation is localized at the shock wave moving through the gas mixture. For a given gas mixture, a deflagration and a detonation may both release the same amount of energy, but the peak pressure of the detonation shock wave is approximately double that of the deflagration and the reflection of the shock wave can produce pressures more than twice as high as those in the shock wave itself (Zalosh 1988). An additional concern is that venting cannot effectively reduce the short-duration peak pressure of a detonation. The response time of the enclosure structure also affects the relative damage produced by deflagrations and detonations.

While the possibility of deflagration in an open space (such as the headspace of a waste tank) is defined purely by the composition of the gas it contains, the possibility of detonation is defined both by the gas composition and the dimensions of the space. In general, either a high length-to-diameter ratio is needed to allow a deflagration to make the transition to detonation or a very high ignition energy is needed to directly produce a detonation.

It is possible for both deflagrations and detonations to be transmitted within the pores of a porous medium or from an open space to the pores. (For assessing Hanford waste, "pores" denotes not only pores between particles but also other larger cracks and bubble networks.) Propagation of detonation in pores, like flame propagation, requires that the pore diameter be greater than some minimum value. The minimum pore diameter is larger (more constraining) for detonation than for deflagration. However, to a large extent, the question of in-pore propagation is moot because most Hanford wastes do not consist of porous media in the sense of a globally interconnecting network of passages but of separated bubbles or small isolated networks.

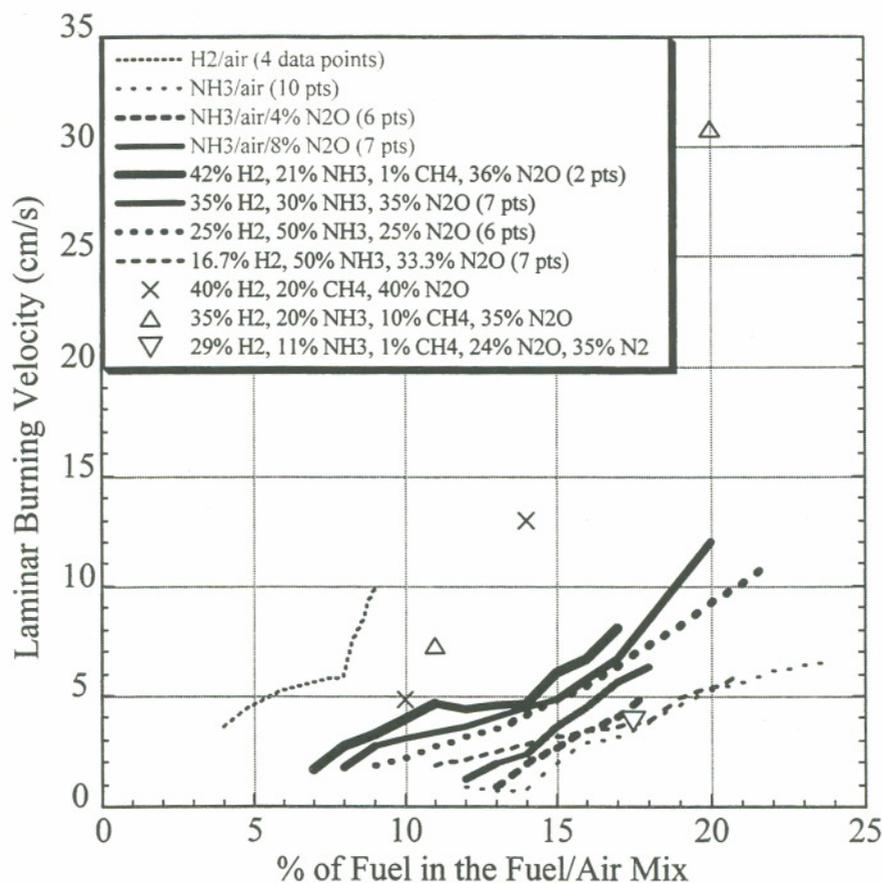
### 4.1 Deflagration in Open Spaces

The potential for deflagration in the open spaces of a tank, such as the headspace itself and the volumes of risers and other tank apparatus, is determined by the flammability of the gas mixture and its associated minimum ignition energy. Both the flame speeds and the peak pressures generated by deflagration are significant in defining the effect of a deflagration on the tank waste and structure.

Flame propagation into a quiescent mixture begins as a laminar spherical front, but turbulence, whether induced by the motion of the front itself or by other gas motion, increases

both the burning velocity and the surface area of the front. Studies have measured the laminar burning velocity for lean waste gas/air mixtures including a number of gas mixtures relevant to the Hanford waste retained gases.<sup>(a,b)</sup> There were few burning velocity data for the fuel-rich range and no data for mixtures in which N<sub>2</sub>O was the only oxidant. Figure 4.1 summarizes the experimental results.

The data show that the burning velocity generally increases as the fraction of fuel in the gas increases. An increase in the amount of H<sub>2</sub> in the waste gas proportionally increases the burning velocity in the lower half of the fuel-fraction range shown in Figure 4.1. This effect can be seen by comparing the behavior of H<sub>2</sub> alone with that of the mixtures containing varying amounts of H<sub>2</sub>. Methane, when present as 10 mol% or more of the waste gas, apparently also increases the



**Figure 4.1.** Laminar Burning Velocities for Gas/Air Mixtures

(a) Ross MC and JE Shepherd. July 1996. *Lean Combustion Characteristics of Hydrogen-Nitrous Oxide-Ammonia Mixtures in Air, Part I*. Report FM96-4, Explosion Dynamics Laboratory, Pasadena, California.

(b) Pfahl U and JE Shepherd. July 1997. *Flammability, Ignition Energy, and Flame Speeds in NH<sub>3</sub>-H<sub>2</sub>-CH<sub>4</sub>-N<sub>2</sub>O-O<sub>2</sub>-N<sub>2</sub> Mixtures*. Report FM97-4R1, Explosion Dynamics Laboratory, Pasadena, California.

burning velocity, though this high a methane concentration has never been measured in Hanford wastes. As the waste gas fraction increases,  $\text{NH}_3$  and mixtures containing primarily  $\text{NH}_3$  do not show as pronounced an increase in burning velocity as the lower- $\text{NH}_3$ , higher- $\text{H}_2$  mixtures. It is also clear that the burning velocity of  $\text{NH}_3$  is higher in the presence of  $\text{N}_2\text{O}$  than in pure air, and that high  $\text{N}_2$  (as in the last mixture listed in the figure's legend) tends to reduce the burning velocity.

The cited references contain a few measurements that indicate that higher burning velocities than those shown in Figure 4.1 are possible. When 75% of the mixture comprising 40%  $\text{H}_2$ , 20%  $\text{CH}_4$ , and 40%  $\text{N}_2\text{O}$  was burned in a fuel-rich regime in air, the burning velocity was 45 cm/s. Turbulent burning is another situation in which high burning velocities were seen. For the same implausibly high- $\text{CH}_4$  mixture, lower-bound turbulent burning velocities of 60 cm/s were measured at fuel fractions of about 25 and 50%. At the lean end of the flammability range (15% fuel and less), the turbulent velocities were about double the laminar velocities. At the rich end (75% fuel), the turbulent velocity was not much greater than the laminar—50 cm/s compared with 45 cm/s.

The measured peak pressure from a deflagration is often compared with the pressure predicted by assuming adiabatic, isochoric<sup>(a)</sup> complete combustion (the AICC pressure). In general, measured peak pressures<sup>(b,c)</sup> were nearly equal to the AICC pressure when the fuel concentration was above the downward propagation LFL and were low when the fuel concentration was at the upward propagation LFL (Slezak et al. 1998). The increase in peak pressure with fuel concentration followed an S-shaped curve. The peak pressures were less than 10 bar in all measurements made with air as the primary oxidant, where the fuel gases included  $\text{H}_2$ ,  $\text{NH}_3$ , and  $\text{CH}_4$ . Peak pressures of about 12 bar were measured for low concentrations of  $\text{NH}_3$  (8 mol% or less) in  $\text{N}_2\text{O}$  with no air present.

In  $\text{H}_2$ - $\text{N}_2\text{O}$ -air mixtures near the downward propagation LFL, the peak pressures remained equal to about 85% of the AICC pressure for  $\text{H}_2$ -air mixtures, though as much as 18%  $\text{N}_2\text{O}$  was added. As was already noted in Section 3.1,  $\text{N}_2\text{O}$  does not react completely in  $\text{H}_2$ -air mixtures until  $\text{H}_2$  reaches about 10 mol%.

In  $\text{NH}_3$ -air mixtures, the peak pressures under turbulent conditions were about 85% of the AICC pressures for  $\text{NH}_3$  concentrations above 16 mol%. However, under quiescent conditions the peak pressures followed the typical S-shape. They were only a little above the initial pressure for 18 mol%  $\text{NH}_3$  or less, while the peak pressures were about 60% of AICC pressure for  $\text{NH}_3$  of 20 mol% or more. As with the burning velocity, the peak pressure is substantially increased by turbulence (in this case, produced by a fan). Like turbulence,  $\text{N}_2\text{O}$  addition caused a decrease in the  $\text{NH}_3$  concentration at which higher peak pressures appeared in combustion of  $\text{NH}_3$ -air mixtures (that is, the rise in the "S" curve occurred at lower  $\text{NH}_3$ ).

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(a) isochoric = constant volume.

(b) Ross MC and JE Shepherd. July 1997. *Lean Combustion Characteristics of Hydrogen-Nitrous Oxide-Ammonia Mixtures in Air, Part I*. Report FM96-4, Explosion Dynamics Laboratory, Pasadena, California.

(c) Pfahl U and JE Shepherd. July 1997. *Flammability, Ignition Energy, and Flame Speeds in  $\text{NH}_3$ - $\text{H}_2$ - $\text{CH}_4$ - $\text{N}_2\text{O}$ - $\text{O}_2$ - $\text{N}_2$  Mixtures*. Report FM97-4R1, Explosion Dynamics Laboratory, Pasadena, California.

Turbulence had much the same effect on H<sub>2</sub>-N<sub>2</sub>O-NH<sub>3</sub> mixtures combined with air as on combinations of single fuel gases and air. Under turbulent conditions in a mixture of 42% H<sub>2</sub>, 21% NH<sub>3</sub>, 36% N<sub>2</sub>O, and 1% CH<sub>4</sub>, the peak pressure rose from the initial pressure at 7 vol% gas to about 90% of the AICC pressure at 15 mol% gas (and remained at that fraction of the AICC pressure for higher gas concentrations). Under quiescent conditions, the same gas mixture remained at the initial pressure even at a concentration of 15 mol% gas, then rose to about 80% of the AICC pressure at 17 mol% gas. Turbulence decreased the steepness of the slope in the "S" curve and moved it to lower gas concentrations. Similar turbulence effects were exhibited by a number of other mixtures that contained H<sub>2</sub>, N<sub>2</sub>O, and NH<sub>3</sub>.

The magnitudes of peak pressures under turbulent conditions can be summarized in the following way. The peak pressures were less than 10 bar in all measurements made with more than 80 vol% air, where the fuel gases included H<sub>2</sub>, NH<sub>3</sub>, and CH<sub>4</sub>. Peak pressures of 12 to 13 bar were measured for low concentrations of NH<sub>3</sub> (8 mol% or less) in N<sub>2</sub>O with no air present. Lean CH<sub>4</sub>/N<sub>2</sub>O mixtures (less than 8 vol% CH<sub>4</sub>) also produced peak pressures in the 12 to 13 bar range. When large amounts of N<sub>2</sub> (60% or more) were added to lean NH<sub>3</sub>/N<sub>2</sub>O and CH<sub>4</sub>/N<sub>2</sub>O mixtures, the peak pressure was reduced by about 30%, depending on how much N<sub>2</sub> was present. A mixture that contained 35 mol% H<sub>2</sub>, 20 mol% NH<sub>3</sub>, 35 mol% N<sub>2</sub>O, and 10 mol% CH<sub>4</sub> produced a peak pressure of 4 bar at 12 vol% in air, 10 bar at 50 vol% in air, and 11 bar in the absence of air. These peak pressures were 80 to 90% of the AICC pressures. A partially inerted mixture that contained 29 mol% H<sub>2</sub>, 11 mol% NH<sub>3</sub>, 24 mol% N<sub>2</sub>O, 1 mol% CH<sub>4</sub>, and 35 mol% N<sub>2</sub> produced a peak pressure of 4 bar at about 22 vol% in air and 9 bar at 90 vol% in air. It was not flammable in the absence of air.

In general, measurements showed that it was typical for the peak pressure to be 90% or less of the AICC values and to be greater than four times the initial pressure. The presence of turbulence increased the peak pressure. The presence of N<sub>2</sub>O also increased peak pressures while inerting with N<sub>2</sub> produced significant reductions in peak pressure.

## 4.2 Deflagration Within Waste

As detailed by Stewart et al. (1996), the gas retained by Hanford waste is present in several different forms:

- small bubbles or bubble/solid aggregates in the liquid in convective layers
- particle-displacing bubbles that may be isolated (in low-strength wet waste) or connected in networks of limited extent (in high-strength wet waste)
- pore-filling bubbles, again in networks of limited extent, probably only present in wet waste with pore diameters on the order of 100 μm
- pores at the top of dry waste that are primarily air-filled but through which is diffusing gas generated in the lower, wetter waste.

Combustion propagation is not considered possible for most of these forms of retained gas. Gas in the convective layer is not capable of combustion, being surrounded by liquid. The same is probably true for isolated particle-displacing bubbles. The possibility of deflagration in dry

waste pores that are open to the air, and therefore contain a gas-air mixture, has been considered and shown to be implausible. Stewart et al. (1996) showed that, in Hanford waste, even with conservatively high gas generation rates, diffusion dilutes the gas in open pores far below flammability. Slezak et al. (1998) demonstrated that it was not possible for waste to both retain gas at potentially flammable concentrations and have pores large enough to support flame propagation. The large pores (high permeability) would allow gas to escape by diffusion, reducing the gas concentration in the pores below flammable levels. In addition, the burning velocities might well be lower in gas-air mixtures than in waste gas alone, making the propagation of flame more difficult in pores that were open to the air. Since the gas in the pores in dry waste will not support combustion, deflagrations cannot propagate from the headspace into the waste.

Depending on the self-flammability of the gas, the diameter of the passages in the bubble network, and the availability of an ignition source, the gas in submerged bubble networks might in theory support combustion. However, the extent of theoretically possible burn propagation is small; waste strength and hydrostatic considerations limit the extent of bubble networks to 1 m or less vertically, and no more than thrice their height horizontally (Stewart et al. 1996). "Fracture bubbles" may exist in the region between the lithodendritic lower waste and the hydrodendritic upper waste. The maximum height of these bubbles is on the order of centimeters (depending on waste strength), and their maximum diameter is believed to be less than 10 times the height.

Studies of pore space deflagration have included experiments with flame propagation in tubes as well as experiments in porous media.<sup>(a)</sup> The experiments in porous media have been conducted with particles of uniform size and shape in the absence of water (which is expected to mitigate combustion). The tank wastes are wet, and the passages are of a range of sizes and shapes. Because of these differences, and because experimental gases have typically been hydrocarbon-air mixtures rather than H<sub>2</sub>-NH<sub>3</sub>-N<sub>2</sub>O-N<sub>2</sub> mixtures, there is uncertainty and probably considerable conservatism in application of these experimental results to tank waste modeling.

The theoretical potential for deflagration in the pore space of a porous medium is defined not only by the flammability of the gas and the ignition energy but also by the Peclet number. The Peclet number is defined as

$$Pe = \frac{S_L c_p \rho d_m}{k} \quad (4.1)$$

where  $S_L$  is the laminar burning velocity of the gas,  $d_m$  the equivalent pore diameter,  $c_p$  the specific heat capacity of the gas,  $\rho$  the density of the gas, and  $k$  the thermal conductivity of the gas.<sup>(a)</sup> For any flammable gas mixture there is a critical Peclet number (pore diameter) below which propagation of the deflagration is not possible. The minimum Peclet number for flame propagation (critical Peclet number),  $Pe^*$ , through a tube was found experimentally to be 85

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(a) Pfahl U, JE Shepherd, and C Unal. February 23, 1998. *Combustion Within Porous Waste*. Report FM97-18, Explosion Dynamics Laboratory, Pasadena, California.

(Lyamin and Pinaev 1985). Experiments conducted by Trimis and Durst (1996) in a dry porous medium found  $Pe^* = 65$ .

Using a critical Peclet number of 65, standard properties for the constituent gases, and a range of laminar burning velocities, we obtained the estimates of minimum equivalent pore diameter for flame propagation that are given in Table 4.1. The burning velocity is uncertain for waste gas mixtures in the absence of air: a lower end value of 10 cm/s can be seen in the velocities measured at moderate concentrations of fuel in air, as in Figure 4.1. Other references have used a burning velocity of 100 cm/s as a basis for estimates.<sup>(a)</sup>

**Table 4.1.** Minimum Pore Diameter to Allow Deflagration in Pores

Source of Gas	$k/c_p\rho$ ( $\text{cm}^2/\text{s}$ @ $40^\circ\text{C}$ )	Minimum Equivalent Pore Diameter (mm)	
		for $S_L = 10$ cm/s	for $S_L = 100$ cm/s
A-101 nonconvective	0.34	22	2.2
AN-103 crust	0.28	18	1.8
AN-103 nonconvective	0.28	18	1.8
AN-104 nonconvective <sup>(a)</sup>	0.19 <sup>(a)</sup>	12 <sup>(a)</sup>	1.2 <sup>(a)</sup>
AN-105 nonconvective	0.24	16	1.6
AW-101 nonconvective	0.19	12	1.2
AX-101 nonconvective <sup>(a)</sup>	0.28 <sup>(a)</sup>	18 <sup>(a)</sup>	1.8 <sup>(a)</sup>
BY-109 nonconvective	0.21	14	1.4
S-102 nonconvective <sup>(a)</sup>	0.15 <sup>(a)</sup>	10 <sup>(a)</sup>	1.0 <sup>(a)</sup>
S-106 nonconvective	0.27	17	1.7
S-111 nonconvective	0.29	19	1.9
SX-106 nonconvective <sup>(a)</sup>	0.20 <sup>(a)</sup>	13 <sup>(a)</sup>	1.3 <sup>(a)</sup>
SY-101 crust <sup>(a)</sup>	0.18 <sup>(a)</sup>	12 <sup>(a)</sup>	1.2 <sup>(a)</sup>
U-103 nonconvective <sup>(a)</sup>	0.14 <sup>(a)</sup>	9.0 <sup>(a)</sup>	0.90 <sup>(a)</sup>
U-109 nonconvective	0.15	10	1.0

(a) Denotes gas mixtures that were marked as potentially self-flammable in Table 3.3. However, the total fuel/oxidant ratio may be too large to allow combustion for AN-104 and AX-101 waste gas.

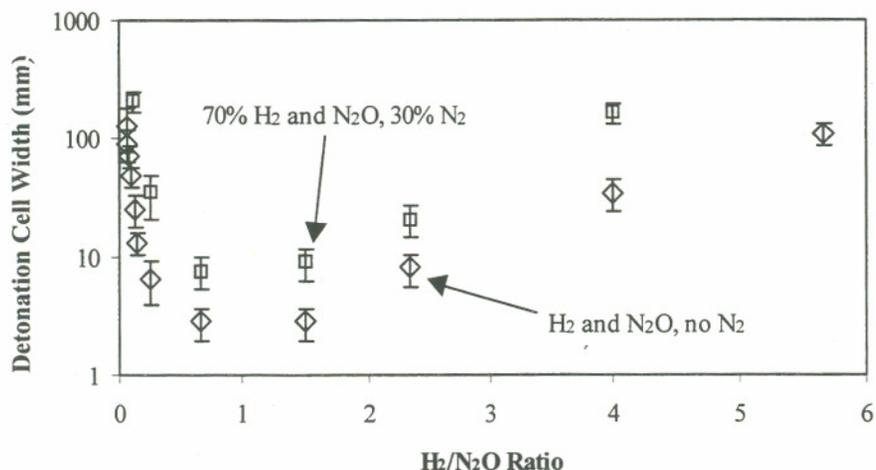
The pores must be greater than about 1 mm in equivalent diameter for the potentially self-flammable gases to deflagrate in the pores. The presence of fracture bubbles or bubble networks in waste would increase the equivalent pore size above what would be expected from the typical particle size of 0.1 mm or less. Based on the current understanding of retained gas morphology, flame propagation within the waste in a region of limited extent cannot be ruled out. Such a region might comprise a bubble network or a fracture bubble with a horizontal extent of as much as a meter.

### 4.3 Detonation in Headspace

The potential for detonation in the open spaces of a tank, such as in the headspace itself and the volumes of risers and other tank apparatus, is defined by the detonation cell width. The larger the cell width, the less the potential for detonation. The detonation cell width,  $\lambda$ , is the measured transverse dimension of the cells generated by the wave structure at the detonation

front and reliably characterizes the detonability of a gas mixture. The cell width is typically measured from the pattern left on a sooted foil by a detonation.

The detonation cell width depends on the properties of the gas. It decreases as the initial pressure increases and varies along a U-shaped curve as the fuel/oxidizer ratio increases. The cell width is typically high near the lower and upper flammability limits and reaches a much lower minimum near the stoichiometric fuel/oxidizer ratio, as shown in Figure 4.2.<sup>(a)</sup> The effect of diluents such as N<sub>2</sub> is to increase  $\lambda$  throughout the range of fuel/oxidizer ratios and narrow the range of fuel/oxidizer ratios for which detonation is possible. Cell widths usually increase monotonically as diluent increases.



**Figure 4.2.** Typical Variation of Detonation Cell Widths with Composition

Detonation cell width measurements have been made for a variety of gas mixtures similar to those in the gases retained in Hanford waste tanks.<sup>(a)</sup> The measurements were made at initial pressures of 0.7 to 1 atm using an extremely high ignition energy of 10 kJ or more. The results, which have been adjusted to 1 atm pressure by Slezak et al. (1998), are shown in Table 4.2.

As shown in Table 4.2, the mixtures representing SY-101 gas tended to maintain constant, relatively small  $\lambda$  values for air in the range from 0% to 40% air. When air was further increased, the cell width increased dramatically. This behavior occurred because the SY-101 mixtures were fuel-rich. The sparseness of measurements for the gas mixtures representing A-101, the three AN tanks, and AW-101 leaves it unclear whether they would show similar behavior.

The SY-101 mixtures gave substantially larger values of  $\lambda$  than the H<sub>2</sub>/N<sub>2</sub>O mixtures for similar amounts of dilution with air. The increase in cell width probably resulted from the presence of NH<sub>3</sub> and (to a lesser extent) of CH<sub>4</sub>. The cell width was considerably greater at low dilution for the SY-101 mixture with N<sub>2</sub> than for the mixture without N<sub>2</sub>, indicating the effect of

(a) Pfahl U, E Schultz, and JE Shepherd. April 10, 1998. *Detonation Cell Width Measurements for H<sub>2</sub>-N<sub>2</sub>O-N<sub>2</sub>-O<sub>2</sub>-CH<sub>4</sub>-NH<sub>3</sub> Mixtures*. Report FM98-5, Explosion Dynamics Laboratory, Pasadena, California.

waste gas N<sub>2</sub>. The tendency of N<sub>2</sub> to reduce detonability (i.e., to increase  $\lambda$ ) can also be seen in the results for the mixture representing AW-101.

From the data presented in Table 4.2 as well as other data,<sup>(a)</sup> the following general conclusions were drawn:

- Cell widths for H<sub>2</sub> and CH<sub>4</sub> were slightly smaller for O<sub>2</sub> as the oxidizer than for N<sub>2</sub>O. However, when NH<sub>3</sub> was the fuel, N<sub>2</sub>O as oxidizer produced cell widths that were a factor of two smaller than for O<sub>2</sub>.
- At stoichiometric fuel/oxidizer ratios, CH<sub>4</sub> produced cell widths that were two to four times larger than produced by H<sub>2</sub>, and NH<sub>3</sub> cell widths were a factor of 10 or more larger than those for H<sub>2</sub>. Additions of as little as 3% CH<sub>4</sub> or NH<sub>3</sub> significantly increased the cell widths over the value found for H<sub>2</sub>/N<sub>2</sub>O mixtures.
- Air dilution produced slightly smaller cell widths in H<sub>2</sub> and CH<sub>4</sub> mixtures than N<sub>2</sub> dilution, with no apparent difference between air and N<sub>2</sub> dilution for NH<sub>3</sub>.
- Dilution of 65 to 80% air or N<sub>2</sub> was sufficient to give cell widths of 100 mm for all stoichiometric fuel/oxidizer combinations except NH<sub>3</sub>/O<sub>2</sub>, for which 35% dilution with N<sub>2</sub> sufficed. However, fuel-rich mixtures (such as the simulated gases for AN-103 and A-101) had cell widths well below 100 mm when diluted 65% by air.

Detonations can be directly caused by very high ignition energies. The minimum energy to produce a detonation in a stoichiometric mixture of H<sub>2</sub> and air is 5 kJ (Zalosh 1988). Even higher energies are required for non-stoichiometric mixtures, or for the other fuels in the tank waste gas. (The minimum detonation energies for fuels in a N<sub>2</sub>O atmosphere are not known.) Under some circumstances, a deflagration flame can accelerate and become a detonation (referred to as a deflagration-detonation transition, or DDT). This is a more common source of detonations than direct induction. In general, for a DDT to occur in an enclosure, the enclosure must have a high length-to-diameter ratio (30 to 100); however, DDTs have been observed in shorter enclosures, and the prediction of DDTs remains uncertain (Zalosh 1988). The geometry of waste tank headspace would probably not encourage a DDT, except in the case of unusually high turbulence or if the volume was constricted by large installed hardware (e.g., a large number of airlift circulators).

If ignition were to occur inside equipment that provided a high length-to-diameter ratio, a DDT could result. The detonation in the equipment could not propagate into the headspace unless the piping in the equipment had a diameter greater than the critical diameter  $d_c$ . According to Zalosh (1988), the critical diameter for this configuration is

$$d_c = 13\lambda \quad (\text{propagate from pipe to open space}) \quad (4.2)$$

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(a) Pfahl U, E Schultz, JE Shepherd. April 10, 1998. *Detonation Cell Width Measurements for H<sub>2</sub>-N<sub>2</sub>O-N<sub>2</sub>-O<sub>2</sub>-CH<sub>4</sub>-NH<sub>3</sub> Mixtures*. Report FM98-5, Explosion Dynamics Laboratory, Pasadena, California.

**Table 4.2. Detonation Cell Widths for Various Dilutions**

Mixture	Diluent	Condition	Dilution	Detonation Cell Width (mm)
stoichiometric H <sub>2</sub> /N <sub>2</sub> O (1:1)	air	No dilution	0%	0.9–1.9
			50%	5–11
		Maximum measured dilution	76%	92–122
stoichiometric H <sub>2</sub> /N <sub>2</sub> O (1:1)	N <sub>2</sub>	No dilution	0%	0.9–1.9
			50%	10–18
		Maximum measured dilution	70%	100–335
42% H <sub>2</sub> , 21% NH <sub>3</sub> , 36% N <sub>2</sub> O, 1% CH <sub>4</sub> (SY-101 without N <sub>2</sub> )	air	No dilution	0%	7–11
			50%	10–18
		Maximum measured dilution	75%	75–181
29% H <sub>2</sub> , 11% NH <sub>3</sub> , 24% N <sub>2</sub> O, 1% CH <sub>4</sub> , 35% N <sub>2</sub> (SY-101)	air	No dilution	0%	19–35
			41%	16–39
		Maximum measured dilution	60%	92–108
31% H <sub>2</sub> , 0.02% NH <sub>3</sub> , 4.3% N <sub>2</sub> O, 1.6% CH <sub>4</sub> , 63% N <sub>2</sub> (AW-101)	air		42%	68–106
47% H <sub>2</sub> , 0.02% NH <sub>3</sub> , 19% N <sub>2</sub> O, 0.9% CH <sub>4</sub> , 33% N <sub>2</sub> (AN-104)	air		40%	8–14
			54%	9–17
61% H <sub>2</sub> , 0.05% NH <sub>3</sub> , 3.8% N <sub>2</sub> O, 0.01% CH <sub>4</sub> , 35% N <sub>2</sub> (AN-103)	air		58%	5–16
			65%	19–42
63% H <sub>2</sub> , 0.02% NH <sub>3</sub> , 11% N <sub>2</sub> O, 0.7% CH <sub>4</sub> , 25% N <sub>2</sub> (AN-105)	air		54%	7–14
			59%	7–14
75% H <sub>2</sub> , 2.4% NH <sub>3</sub> , 5.6% N <sub>2</sub> O, 0.7% CH <sub>4</sub> , 16% N <sub>2</sub> (A-101)	air		62%	8–14
			68%	7–20

#### 4.4 Detonation Within Waste

The potential for detonation in the pore space of tank waste is defined by the detonation cell width of the gas itself and by the pore diameter. The same physical arguments that made it impossible for pores that were open to the air to hold enough gas to support deflagration also apply to detonation (Slezak et al. 1998). Therefore, the gas must be capable of supporting detonation under airless conditions

As was the case for deflagrations, detonations cannot propagate within pores or from an open space into pores when the pore diameter is less than a certain critical value. The critical (minimum) diameter has been assumed<sup>(a)</sup> to be the same as for propagation from a pipe into an open space,  $d_c > 13\lambda$  (Eq. 4.2).

(a) Pfahl U, JE Shepherd, and C Unal. February 23, 1998. *Combustion Within Porous Waste*. Report FM97-18, Explosion Dynamics Laboratory, Pasadena, California.

Of the mixtures that were used to simulate gas retained in tank waste (see Table 4.2 for a list), only one had the detonation cell width measured at 0% air dilution. This SY-101 gas simulant (29% H<sub>2</sub>, 11% NH<sub>3</sub>, 24% N<sub>2</sub>O, 1% CH<sub>4</sub>, 35% N<sub>2</sub>) is somewhat less fuel-rich than the retained gas that was subsequently measured by RGS (34% H<sub>2</sub>, 19% NH<sub>3</sub>, 19% N<sub>2</sub>O, 0.6% CH<sub>4</sub>, 27% N<sub>2</sub>). The cell width for the retained gas could therefore be somewhat smaller than that measured for the simulant, 19 to 35 mm (Table 4.2). However, the critical diameter for propagation (260 to 460 mm) is so much greater than the expected size of pores, networks, or fractures in the waste that a slightly smaller critical diameter would not change the conclusion that detonation cannot propagate within the SY-101 matrix.

A reaction model, calibrated using the cell widths measured at high dilution,<sup>(a)</sup> was used to estimate the cell widths for undiluted mixtures representing gases in A-101, AN-103, AN-104, AN-105, and AW-101. The results showed that there was too little N<sub>2</sub>O in the gas simulants for AW-101, AN-103, and A-101 to allow a detonation in the undiluted gas. In the AN-104 mixture (19% N<sub>2</sub>O) the cell width was calculated to be about 80 mm, and, in the AN-105 mixture (11% N<sub>2</sub>O), 4000 mm. Here again, the critical diameter 13λ for detonation propagation is much higher than any plausible passage size.

Of the tanks whose gases were not simulated, several (according to Table 3.3) retained gas that contained H<sub>2</sub> and N<sub>2</sub>O that were comparable to the gas simulants for AN-104 and AN-105. This group included AX-101 (60% H<sub>2</sub>, 11% N<sub>2</sub>O), BY-109 (50% H<sub>2</sub>, 18% N<sub>2</sub>O), S-106 (63% H<sub>2</sub>, 11% N<sub>2</sub>O), and S-111 (66% H<sub>2</sub>, 11% N<sub>2</sub>O). It seems likely that cell widths at 0% air dilution would be at least 50 to 100 mm for these gases, though this estimate is uncertain and a reaction zone model would be needed to provide a better estimate. Part of the uncertainty comes from the compositions of the gases, part from the fact that the gases are retained at hydrostatic pressures of 1.5 atm or more. The higher initial pressure would be expected to reduce the cell width, compared to 1-atm measurements.

Tanks S-102 (33% H<sub>2</sub>, 33% N<sub>2</sub>O), SX-106 (51% H<sub>2</sub>, 24% N<sub>2</sub>O), U-103 (23% H<sub>2</sub>, 40% N<sub>2</sub>O), and U-109 (25% H<sub>2</sub>, 27% N<sub>2</sub>O) contained much more N<sub>2</sub>O and more closely resembled the gas simulant for SY-101. The SY-101 gas simulant had a cell width of 19 to 35 mm at airless conditions and 1 atm. Again, using this cell width for other (though similar) compositions introduces much uncertainty. However, because the critical diameter is 13λ, even the minimum λ measured for nitrogen-containing gas mixtures (about 7.5 mm, for a mixture of 28% H<sub>2</sub>, 42% N<sub>2</sub>O, 30% N<sub>2</sub>, similar to U-103 gas) still implies that a crack of 98 mm (about 4 in.) diameter would be required for detonation propagation.

Not only would a crack in waste need to be of a large diameter throughout its length, but the length of the crack would probably need to be at least 10 times the diameter for a detonation to make the transition to a detonation. Our conclusion is that detonation propagation is not plausible in the tanks whose retained gas compositions have been measured.

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(a) Pfahl U, JE Shepherd, and C Unal. February 23, 1998. *Combustion Within Porous Waste*. Report FM97-18, Explosion Dynamics Laboratory, Pasadena, California.

## 4.5 Conclusions

The following conclusions were drawn from the deflagration and detonation research summarized above:

- When the fuel concentration in air was at or above the downward propagation LFL, deflagrations of waste gas mixtures typically produced peak pressures of 80% to 90% of the pressure predicted for adiabatic, isochoric complete combustion. The AICC pressure can therefore be used as a slightly conservative representation of actual pressures resulting from a burn.
- Peak pressures from deflagrations of waste gases were usually four or more times the initial pressure and were less than 10 times the initial pressure in all the measurements made with air as the primary oxidant.
- The presence of  $N_2O$  increased the peak pressures of deflagrations, while  $N_2$  significantly reduced the peak pressure.
- Deflagrations are unlikely to propagate within Hanford wastes because retained gas does not appear to take the form of millimeter-diameter pores interconnected in a large network. Creating an ignition source within the waste is also problematic. However, small-scale deflagrations involving fracture bubbles of several cm extent or bubble networks of up to 1 m extent cannot be ruled out.
- $NH_3$  and  $CH_4$  reduced the detonability of  $H_2/N_2O$  mixtures.
- Dilution of 65 to 80% air or  $N_2$  was sufficient to give detonation cell widths of 100 mm for all stoichiometric fuel/oxidizer combinations except  $NH_3/O_2$ , for which 35% dilution with  $N_2$  sufficed. However, fuel-rich mixtures (such as the simulated gases for AN-103 and A-101) had cell widths well below 100 mm when diluted 65% by air.

Detonations will not propagate within the RGS-sampled Hanford wastes; the same conditions that make deflagration unlikely make detonation implausible.



## 5.0 Ignition

The preceding sections show that some of the gas mixtures created within the waste and released to the headspace of the tank are indeed flammable. If these gases were allowed to collect without dilution, ignition could result in a deflagration or detonation. This section evaluates the requirements for initiating a burn of flammable gases.

### Requirements for Ignition

Mechanisms for ignition of a given mixture of combustible gases are complex and involve formation of a critical concentration of molecular fragments by the energy in electrostatic or mechanical sparks. Several energy sources and configurations have been used to study flame ignition. It has been determined that an ordinary capacitor discharge spark consistently yields the lowest ignition energy for any specific combustible mixture (Strehlow 1984). Therefore, experiments that use capacitive spark discharge may (conservatively) underestimate the ignition energy that would be required for ignition if supplied by less effective energy sources.

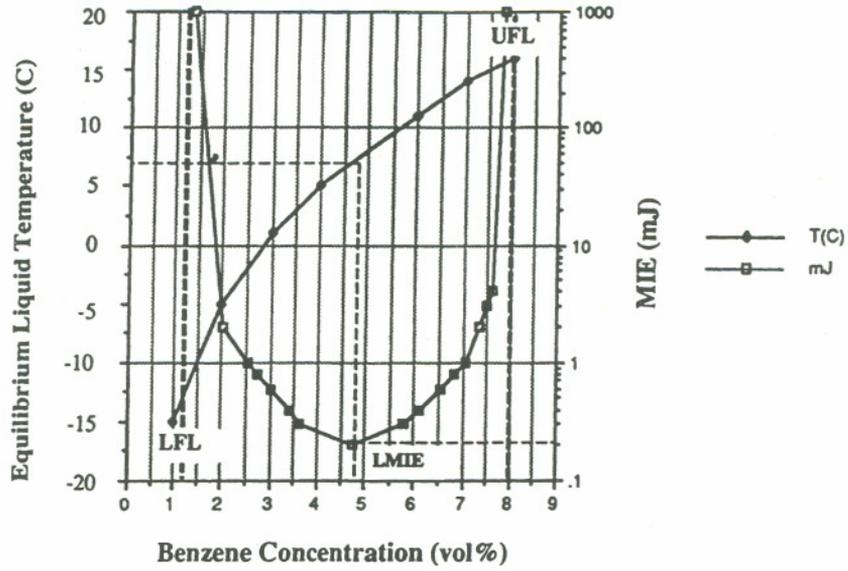
Studies of the requirements for ignition of hydrogen have defined the “minimum ignition energy” (MIE), the energy below which the ignition of a combustible mixture cannot occur and above which ignition occurs. Most of the work on minimum ignition energies was done with capacitive spark discharges. For hydrogen the MIE is on the order of 0.01 mJ (Pratt 1997). To put this in perspective, the range of energies in sparks from static discharges perceived by humans can range from microjoules to tens of millijoules. A 1 mJ spark is perceptible, a 10 mJ spark is a prick, a 30 mJ spark is a sharp prick, and a 100 mJ spark will result in a slight jerk (Pratt 1997). Thus, the sparks from humans touching metal far exceed the energy required to ignite some hydrogen/air mixtures.

The MIE for a gas varies with the stoichiometry of the mixture, as shown for benzene in Figure 5.1 (Britton 1992). The optimum concentration that gives the lowest MIE (LMIE) is not the stoichiometric fuel/oxidant mixture but a slightly fuel-rich one. The MIE at the LFL is several orders of magnitude greater than that for the optimum mixture. This same situation exists for hydrogen-oxygen systems (Lewis and Elbe 1987, p. 347), as shown in Figure 5.2. The MIE is about 0.02 mJ for H<sub>2</sub>-O<sub>2</sub>-N<sub>2</sub> at 25–30% hydrogen. At 5% H<sub>2</sub> the MIE is greater than 1 mJ.

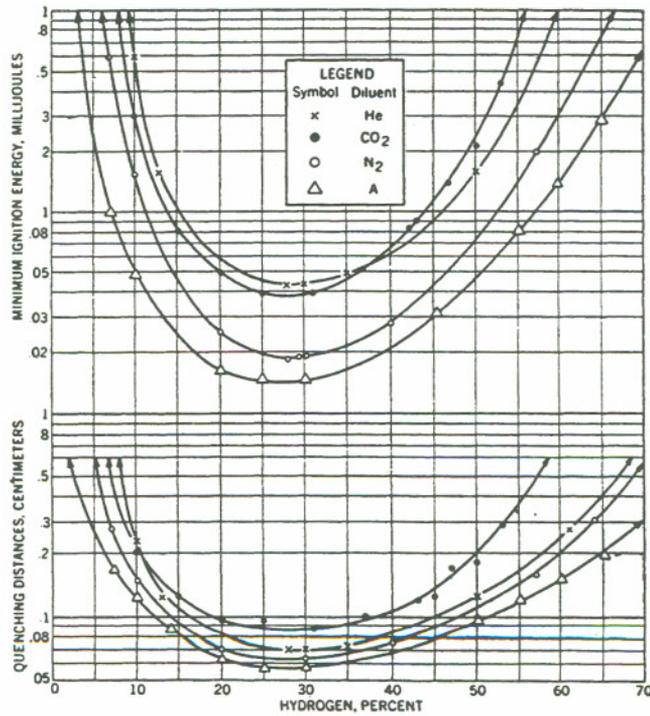
Because the gas mixtures of interest to the Flammable Gas Safety Issue might behave differently than the individual components, an experiment was conducted at the California Institute of Technology<sup>(a)</sup> to evaluate the effect of various ignition energies on the LFL of mixtures relevant to Hanford tank waste gases containing H<sub>2</sub>, NH<sub>3</sub>, N<sub>2</sub>O, CH<sub>4</sub>, and N<sub>2</sub>. The researchers found that, for ignition energies less than 10 J, the flammability limit (in air) of Mixtures 27 (40% H<sub>2</sub>, 40% N<sub>2</sub>O, 20% CH<sub>4</sub>), 28 (35% H<sub>2</sub>, 35% N<sub>2</sub>O, 10% CH<sub>4</sub>, 20% NH<sub>3</sub>), and 29 (29% H<sub>2</sub>, 24% N<sub>2</sub>O, 11% NH<sub>3</sub>, 35% N<sub>2</sub>, 1% CH<sub>4</sub>) occurred at 7%, 9%, and 14% fuel in air, respectively. None of the three mixtures showed any pronounced dependence of the LFL on ignition energy between 0.04 and 8 J.

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(a) Pfahl U and JE Shepherd. 1997. *Flammability, Ignition Energy, and Flame Speeds in NH<sub>3</sub>-H<sub>2</sub>-CH<sub>4</sub>-N<sub>2</sub>O-O<sub>2</sub>-N<sub>2</sub> Mixtures*. Report FM97-4R1, Explosion Dynamics Laboratory, Pasadena, California.



**Figure 5.1.** Typical Relationship Between Stoichiometry and Minimum Ignition Energy (Britton 1992)



**Figure 5.2.** Relationship Between Stoichiometry and Minimum Ignition Energy for  $H_2/O_2/Inert$  (Lewis and Elbe 1987)

Friction sparks that result from the impact of two surfaces can serve as an ignition source for flammable gas mixtures. The temperature of a spark struck from steel can approach 1400°C. The ignition potential of a spark depends upon its total heat content; thus the size of the particle produced is an important factor. Mechanical sparks cool quickly and can ignite gases only under very favorable conditions. A study was done to determine which metal would be best suited for use in Tank 241-SY-101 (Pearce 1992, p. 45-57). The study showed that hot-spot temperatures achieved by normal or glancing blows were less than the temperature required to ignite hydrogen-air mixtures; that the presence of rust made ignition more likely; and that stainless steel was one of the most sparking-resistant materials available. The results of this work led to using stainless steel for new components in Tank 241-SY-101 and other similar tanks. In addition, special tools of copper-beryllium and other low-sparking alloys are being used at the Hanford Site to minimize the danger of sparks in hazardous locations.

Flammable gas mixtures can be ignited by a hot surface. The minimum temperature at which a mixture will burst into flame spontaneously is known as the auto-ignition temperature (AIT). The AIT is typically measured in air. Some typical values are 520°C for hydrogen, 650°C for ammonia, and 630°C for methane (Kuchta 1985). An evaluation was conducted at CIT<sup>(a)</sup> for gas mixtures relevant to the Hanford tanks. For H<sub>2</sub>-N<sub>2</sub>O-NH<sub>3</sub> mixtures, the investigators estimated the AIT to be in the range of 457° to 507°C. Procedural protections restrict welding and other activities that might produce such high surface temperatures.

The dome spaces of the Hanford tanks are connected to the ambient environment via filtered ventilation systems. Due to the water content of the waste and the humidity of the inlet air, the dome spaces have a reasonably high relative humidity; that is, the atmosphere in the headspace is not dry. The relative humidity can range from 20 to 60%. As noted in Section 3.3, the humidity does not have a significant effect on the LFL of gas mixtures in which hydrogen is the main contributor; however, high humidity could be expected to reduce the sparking potential of some materials.

In summary, ignition of flammable mixtures that contain hydrogen as the principal component can occur at very low levels of energy for electrostatic sparks. Accordingly, the operating controls for the tank farms at the Hanford Site incorporate controls for ignition sources and for monitoring flammable gases.

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